

Title (en)
TERMINAL GUIDANCE METHOD AND GUIDED MISSILE USING IT

Publication
EP 0081421 B1 19890125 (FR)

Application
EP 82402180 A 19821130

Priority
FR 8123025 A 19811209

Abstract (en)
[origin: US4568040A] A guidance method is provided for the terminal portion of the trajectory of a guided missile having a sensor and comprising two sections coupled together by a central shaft and free to rotate with respect to one another about the longitudinal axis of the missile; one section comprises a drive means for controlling the roll attitude of this section and a gas generator which feeds a nozzle for providing a transverse throat force and the other section has a stabilizing tail unit formed by a set of fins able to be opened out.

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IPC 8 full level
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F42B 10/661 (2013.01 - EP US); **F42C 13/006** (2013.01 - EP US)

Cited by
DE3323685A1; FR2555821A1; FR2583868A1; US4679748A; FR2568365A1; EP0223919A1; FR2736147A1; FR2684723A1; GB2167536A;
FR2578043A1; FR2634012A1; GB2175377A; FR2581750A1; US4711178A; FR2711783A1; US5564651A; EP0384965A3; EP0191121A1;
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