

Title (en)
SPACECRAFT THRUSTER

Title (de)
ANTRIEBSSYSTEM FÜR RAUMFAHRZEUG

Title (fr)
PROPULSEUR D'ENGIN SPATIAL

Publication
EP 1604111 A4 20060118 (EN)

Application
EP 04809320 A 20040317

Priority

- US 2004008054 W 20040317
- EP 03290712 A 20030320
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Abstract (en)
[origin: EP1460267A1] A thruster has a chamber (6) defined within a tube (2). The tube has a longitudinal axis which defines an axis (4) of thrust; an injector (8) injects ionizable gas within the tube, at one end of the chamber. A magnetic field generator with two coils (12, 14) generates a magnetic field parallel to the axis; the magnetic field has two maxima along the axis (4); an electromagnetic field generator has a first resonant cavity (16) between the two coils generating a microwave ionizing field at the electron cyclotron resonance in the chamber (6), between the two maxima of the magnetic field. The electromagnetic field generator has a second resonant cavity (18) on the other side of the second coil (14). The second resonant cavity (18) generates a ponderomotive accelerating field accelerating the ionized gas. <??>The thruster ionizes the gas by electron cyclotron resonance, and subsequently accelerates both electrons and ions by the magnetized ponderomotive force. <IMAGE>

IPC 1-7
F03H 1/00

IPC 8 full level
F03H 1/00 (2006.01)

CPC (source: EP KR)
B64G 1/40 (2013.01 - KR); **F03H 1/0081** (2013.01 - EP KR)

Citation (search report)

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- [XA] US 6193194 B1 20010227 - MINOVITCH MICHAEL A [US]
- [A] EP 0359732 A2 19900321 - HARRIS BLAKE CORP [US]
- [A] US 3279175 A 19661018 - HENDEL HANS W, et al
- [A] US 6293090 B1 20010925 - OLSON LYNN B [US]
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