

Title (en)
BYPASS TURBOJET ENGINE NACELLE

Title (de)
BYPASSGASTURBINENGONDEL

Title (fr)
NACELLE POUR TURBORÉACTEUR DOUBLE FLUX

Publication
EP 2076438 A2 20090708 (FR)

Application
EP 07858402 A 20071005

Priority

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Abstract (en)
[origin: WO2008043903A2] The nacelle (1) comprises an air intake (5) upstream of the turbojet engine (2), a middle section (6) an internal casing (6a) of which is intended to surround a fan (3) of the turbojet engine (2), and a downstream section (7) comprising an external structure (7a) which is rigidly connected to a downstream part of the casing (6a) of the fan (3) so as to support the turbojet engine (2) and has means of attachment to a strut (12) intended to be connected to a fixed structure (13) of an aeroplane.

IPC 8 full level
B64D 27/18 (2006.01)

CPC (source: EP US)
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