

Title (en)

High temperature components with thermal barrier coatings for gas turbine

Title (de)

Hochtemperaturbauteile mit Wärmedämmenschichten für Gasturbine

Title (fr)

Composants à haute température avec des revêtements de barrière thermique pour turbine à gaz

Publication

EP 2725120 A1 20140430 (EN)

Application

EP 13189647 A 20131022

Priority

JP 2012234278 A 20121024

Abstract (en)

The most principal feature of the present invention is as follows: Namely, in the gas-turbine-use high-temperature component including the thermal barrier coating and a cooling structure, micro passages are provided inside an alloy bond-coat layer (2) and a thermal-barrier ceramic top-coat layer (3) of the thermal barrier coating, the micro passages being in communication from the substrate side to the surface side. Moreover, a partial amount of coolant of a coolant (8) for cooling the high-temperature component is caused to flow out to the outside of the high-temperature component via these micro passages. The employment of the structure like this makes it possible to expect the implementation of a high-temperature component's heat-resistant-temperature enhancement effect based on the transpiration cooling effect.

IPC 8 full level

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Citation (applicant)

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Citation (search report)

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