

Title (en)
INERTIAL MEASUREMENT SYSTEM FOR AN AIRCRAFT

Title (de)
INERTIALMESSUNGSSYSTEM FÜR EIN FLUGZEUG

Title (fr)
SYSTEME INERTIEL DE MESURE POUR AERONEF

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Application
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Abstract (en)
[origin: WO2016207029A1] The invention relates to an inertial measurement system for an aircraft, including at least one processing unit connected to at least two inertial units, namely at least one inertial unit of a first type comprising at least three angular sensors and three linear sensors aligned on three separate sensitive axes, and one inertial unit of a second type, comprising at least four angular sensors and four linear sensors aligned on four separate sensitive axes. The system is arranged to calculate one piece of inertial data from the six sensors of the inertial unit of the first type and four pieces of inertial data from four separate combinations of three angular sensors and three linear sensors selected among the sensors of the inertial unit of the second type, and the processing unit is arranged to compare the four pieces of inertial data from the measurements of the inertial unit of the second type with one another so as to determine whether the inertial unit of the second type is faulty.

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