

Title (en)  
TURBOMACHINE FLOW CHANNEL

Title (de)  
TURBOMASCHINEN-STRÖMUNGSKANAL

Title (fr)  
CANAL D'ÉCOULEMENT DE TURBOMACHINE

Publication  
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Application  
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Abstract (en)  
[origin: US2019169989A1] The present invention relates to a method for designing a flow channel for a turbomachine, in particular a gas turbine that comprises a guide vane cascade having a plurality of guide vanes, which are distributed in the peripheral direction, and flow passages, each of which is bounded by two successive guide vanes, and a support rib arrangement having at least one support rib, wherein a design of one of the flow passages is adapted to this support rib, that it is situated downstream of, in order to reduce a pressure loss and/or a vibrational stimulation.

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