

Title (en)

DAMPER FOR A COMBUSTOR ASSEMBLY OF A GAS TURBINE ASSEMBLY, COMBUSTOR ASSEMBLY COMPRISING SAID DAMPER AND METHOD FOR MANUFACTURING A DAMPER FOR A COMBUSTOR ASSEMBLY

Title (de)

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Title (fr)

AMORTISSEUR POUR ENSEMBLE DE CHAMBRE DE COMBUSTION D'UN ENSEMBLE DE TURBINE À GAZ, ENSEMBLE DE CHAMBRE DE COMBUSTION COMPORTANT LEDIT AMORTISSEUR ET PROCÉDÉ DE FABRICATION D'UN AMORTISSEUR POUR UN ENSEMBLE DE CHAMBRE DE COMBUSTION

Publication

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Application

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Priority

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Abstract (en)

A damper for a combustor assembly (3) of a gas turbine assembly (1) is provided with:at least two damping volumes (32, 35) defined by respective damping bodies (31, 34); the damping volumes (32, 35) being interconnected in fluidic communication;at least one perforated connecting plate (36) connecting the two damping volumes (32, 35);at least one perforated end plate (38) configured to connect at least one of the two damping volumes (32, 35) with a combustion chamber (16; 14) of the combustor assembly (3);the perforated connecting plate (36) and the perforated end plate (38) being provided with a plurality of openings (42);the openings (42) being dimensioned so as to operate in a low Strouhal regime according to the following formula $\omega \cdot RH / Ub < 0,5$ wherein:
• ω is the angular frequency correlated to the frequency to damp according to the following relation $\omega = 2\pi f$;
• R is the equivalent radius of one of the openings (42);
• U_b is the velocity of the flow through one of the openings (42).

IPC 8 full level

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CPC (source: CN EP)

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