

Title (en)
INTERCEPTOR MISSILE AND METHOD FOR STEERING SAME

Title (de)
ABFANGFLUGKÖRPER UND VERFAHREN ZU DESSEN LENKUNG

Title (fr)
MISSILE D'INTERCEPTION ET SON PROCÉDÉ DE GUIDAGE

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Application
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Abstract (en)
[origin: WO2022084068A1] In a method for steering a steerable interceptor missile (2) for intercepting a moving target (8), during a midcourse phase (PM) of the interception, which interceptor missile is driven by an engine (10), the interceptor missile (2) is steered with the aid of real steering commands, which are produced at respective steering times on the basis of free control parameters (SP), which free control parameters are in the form of a current parameter vector (10), the free control parameters (SP) are continually and repeatedly optimized during the midcourse phase (PM) with the aid of an optimization method (14) for optimizing the control parameters (SP), the optimization method (14) is carried out in parallel with the actual steering, newly learned information about the movement of the target (8) and/or information about the flight of the interceptor missile (2) is used in the optimization method (14) as soon as said information is available, and optimized control parameters (SP) are accepted into the current parameter vector (10) after said optimized control parameters are provided by the optimization method (14). An interceptor missile (2) contains the current parameter vector (10) and a control and evaluation unit (50) for carrying out the method according to the invention.

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