(19) Europäisches Patentamt
European Patent Office
Office européen des brevets

(11) Publication number: **0 066 019**
A1

(12) # EUROPEAN PATENT APPLICATION

(21) Application number: **81302453.6**

(22) Date of filing: **03.06.81**

(51) Int. Cl.³: **C 23 C 9/02**

(43) Date of publication of application:
08.12.82 Bulletin 82/49

(84) Designated Contracting States:
DE FR GB NL

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(54) **Diffusion coating composition and method.**

(57) Improved powder-type diffusion-coating compositions for use in codeposition processes involving the formation of diffusion coating of chromium and aluminum on high nickel and high cobalt alloys known as superalloys. The compositions are characterised by easy flowability and, particularly, by the use of the intermetallic compound Co_2Al_9.

EP 0 066 019 A1

(EP12924M)
(1.6.81)

DIFFUSION COATING COMPOSITION AND METHOD

The invention relates to the diffusion coating of superalloys, such as high-nickel and high-cobalt alloys, wherein a part to be treated is placed in a coating, powder pack.

It is already known that diffusion coating of metallic objects, for example nickel - and cobalt-based alloys, may be carried out by embedding the article to be coated in a powdered coating pack including:

(1) an inert filler,
(2) a vaporisable carrier ingredient, and
(3) powdered sources of metal materials to be diffused into the superalloy object.

The vaporisable carrier ingredient, usually a halide composition, acts as a flux in facilitating the initial reaction between aluminum and the alloy being treated, and also acts to accelerate the diffusion process by forming intermediate or transient compounds during the process. In general, the vaporisable carrier material has been a halide, for example a fluoride or chloride

salt, such as ammonium chloride. This relatively diffusable material provides means for carrying the treating metal into the superalloy surfaces to be treated.

The metal powder, usually aluminum or chromium, is the active metal-treating agent on which the carrier material acts to facilitate diffusion of the metal into the article to be treated.

The inert filler acts primarily as a means to moderate the concentration and rate at which the carrier material and diffusing metal approach the metal article to be treated. It is also a manipulative expedient which provides the function of reducing the cost of material expended in carrying out the process. Cost of the material in a powder pack usually requires that the pack is to be used for a number of treating cycles. In such cases the pack is usually refurbished with the more-readily depleted components before proceeding from one cycle to another.

Processes of this general type have been disclosed by Wachtell and Seelig in U.S. Patent 3,257,230 and by Puyear and Schley in U.S. Patent 3,079,276. The object of such processes is to provide a protective outer sheathing on engineering parts subjected to high temperatures and corrosive atmospheres; for example, the turbine blades in jet aircraft engines are subjected to such temperature and environment.

Some additional prior art includes packings which utilise Co_2Al_5 (e.g. PWA 273, a material sold by Pratt & Whitney Division of United Technologies, Inc. and packings such as those described in a number of Patents including U.S. Patents 3,716,398; 3,594,219; 3,577,268; 3,810,782, 4,024,294; 4,041,196; 3,979,274; and many more such patents

primarily classified in U.S. Classification 117 (old) and Class 427.

There are many aspects of these diffusion processes for pack cementation coatings that it would be desirable to enhance. Some such aspects relate to the mechanical properties of the powder composition: for example, it would be very desirable for the used coating powder to be readily released from some of the very small apertures and channels that are encountered in the alloy parts being treated. Moreover, it is always desirable to improve the "hot corrosion" resistance of internal as well as external surfaces of the parts by subjecting them to treatments of the type being described.

As will be described below, the invention described herein is based on the surprising discovery of the value of certain cementatious coating powders.

It is an object of the present invention to provide improved powder compositions for use in cementatious pack coating of super alloys.

A more particular object of the invention is to provide such compositions which form means to improve the properties of parts treated with the new composition.

Another object of the invention is to provide compositions of superior flowability after they have been subjected to processing temperatures.

Other objects of the invention are to obtain an improved diffusion coating process, one which may be used to achieve more rapid processing and one which may be used to achieve superior properties of the processed goods.

Other objects of the invention will be obvious to those skilled in the art on their reading of this disclosure.

The above objects have been substantially unexpectedly and surprisingly achieved by the development and use of a powdered cementation pack coating composition comprising the inter-metallic powder Co_2Al_9, particularly the powder which passes 325 mesh.

The amount of Co_2Al_9 that is utilised in the composition of the invention will differ somewhat depending on the particular use of the coating. For example, different quantities will be used in packings that are intended for internal surfaces (i.e. surfaces of interstices or orifices) and in external surfaces (i.e. the larger exterior surface of a part). One advantage of the compositions of the invention is the fact that they can readily be used in a single coating operation both for internal and external use. Another advantage is that the processes may be carried out at lower time-temperature profiles than processes carried out with such intermetallics as Co_2Al_5, Cr_8Al_5 or CoAl.

The superalloys which are believed to be most advantageously treated by the process of the invention are the nickel-based superalloys especially e.g. those comprising about 16% chromium or less. Particular advantage is achieved with those alloys such as are known as IN100 and IN713 and IN792 which contain about 13% chromium or less.

It will be understood by those skilled in the pertinent art that the compositions of powder-pack compositions are varied depending upon the particular application involved. For example, somewhat higher chromium content in the metal being treated usually dictates a somewhat higher chromium

content in the powder to facilitate diffusion of chromium into the surface of the metal. Likewise, the presence of cobalt in the metal slows diffusion of aluminium and this effect can be counteracted by changing the powder composition. Also, the powder weight to surface area ratio is so different for internal surfaces of a part being treated from powder weight to surface area ratio achieved at exterior surfaces, that this factor dictates substantial changes in coating powder composition. (Thus, for example, the internal surfaces of a part being prepared for treatment could suitably contain a coating powder composed of 8 to 10% Co_2Al_9 and 3 to 4% chrome, whereas the external surfaces being prepared for treatment might be packed with a mixture of 3-4% Co_2Al_9 and 2-4% chrome.)

The process of the invention can typically be accomplished in less than half the time required for similar processing with conventionally used powder packings. For example, a cycle time of about 1 1/2 hours can be used in some applications as opposed to 3 to 4 hour cycles achieved with powder pack compositions which are presently available. Thus a 50 microns thick diffusion coating can be imparted to, say IN100 or IN713 alloy in 1 1/2 hours at a heat treating temperature of 1052°C, whereas the same result would take at least about twice as long using the powder packs of the prior art.

Oxidation resistance of coatings prepared according to the invention may be improved from 100% to 200% over coatings of the prior art, when utilising a 3% Co_2Al_9, 2% chromium powder pack ("3-2 mix"). This has been shown by a high velocity oxidation test which is intended to be a partial simulation of operating conditions in the hot zone of a jet engine. Excellent results are also

achieved with a 3-4 mix. Thus products comprising the diffusion coating of the invention exhibit excellent oxidation/ablation resistance at temperatures of 1093-1149°C.

A simulated Inconel 713 turbine blade coated with a co-deposited diffusion coating derived from a 3% Co_2Al_9 4% Cr pack, shows hot corrosion resistance approximately 50% greater than a "state of the art" aluminide coating. This has been validated by a 943°C - 954°C hot corrosion test which, in part, simulates jet engine operating conditions.

One valuable and novel characteristic of the compositions of the invention is that they are resistant to being immobilized by a heat treatment in Argon at about 2000°F (1093°C) and, therefore, remain readily flowable after two hours of such a treatment.

In this application there is shown and described preferred embodiments of the invention and suggested various alternative and modifications thereof, but it is to be understood that these are not intended to be exhaustive and that other changes and modifications can be made within the scope of the invention. These suggestions herein are selected and included for purposes of illustration in order that others skilled in the art will more fully understand the invention and the principles thereof and will be able to modify it, each as may be best suited in the condition of a particular case.

EXAMPLE 1

A turbine blade, whose service condition has been categorized by a jet engine operator as high oxidation and cast from a high-nickel alloy sold under the trade

designation "IN-100" by the International Nickel Company, was degreased by exposure to trichloroethane solvent vapours. (After this degreasing all handling of the turbine blade was carried out with cotton gloves).

Thereupon the area of the turbine blade to be subjected to the diffusion coating process was abrasively cleaned with Al_2O_3 grit which had passed a 120 mesh sieve but has been retained on a 220 mesh sieve. After this blasting process, the turbine blade was once more degreased.

After these preparatory steps, the turbine blade was packed into a coating container, which has been prepared according to procedures known in the art and packed in a coating powder formulation comprising:

Constituents	% by weight
Calcined aluminium oxide (pass 100 mesh)	94.5%
Co_2Al_9 (pass 325 mesh)	3.0%
Chromium powder (pass 325 mesh)	2.0%
Sodium Fluoride	0.5%

This is designated as the RB-505A blend for applications requiring high oxidation resistance.

Workpieces should be placed in the coating container in such a way that there is about a 12.7mm gap between adjacent pieces.

The powder box was loaded into a retort which is provided with means to circulate gas therethrough, means to insert thermocouples thereinto for the remote reading of temperature therein and a sand seal to prevent the ingress of air thereto. After the retort was closed, it was purged with Argon gas at a rate of about 7 volume changes

per hour and then placed into a gas-fired pit furnace. Argon gas was constantly fed into the retort at a rate of about 5 volume changes per hour as the temperature inside the retort was rapidly raised to 1052°C and held there for an hour. The retort was then withdrawn from the furnace, and the parts were unpacked from the powder pack.

The coated nickel-base turbine blades were carefully cleaned with a stiff-bristled brush and compressed air. Thereupon, the part was inspected and washed for three minutes in warm water and dried.

The parts were then loaded in a clean retort not previously used for diffusion coating and heat treated in a hydrogen atmosphere for one hour at 1066°C. Purging technique and gas flow rates were similar to that described for the coating process, above.

After metallographic examination of a test piece so treated, an excellent codeposited diffusion coating of about 63.5 microns on depth was achieved during this process.

EXAMPLE 2

Example 1 was repeated but now a turbine nozzle guide vane of Inconel 738 alloy whose service condition has been categorised by a jet-engine operator as high hot corrosion. The following powder formulation was used:

Constituents	Parts by Weight
Co_2Al_9 ≠ 325 mesh	3.0
Chromium, # 325 mesh	4.0
NaF	0.5
Calcined aluminum oxide, # 100 mesh	92.5

This is designated as the RB505-B blend for applications

requiring high hot corrosion resistance.

The pack temperature was 1066°C and the treatment time was two hours in an Argon atmosphere. The post-treatment was at 1079°C for one hour in a hydrogen atmosphere and resulted in an excellent codeposited diffusion coating of aluminum and chrome of 63.5 microns in depth.

EXAMPLE 3

This example relates to a hollow-turbine blade with internal cooling passages.

Example 1 was repeated excepting that the parts being treated had small apertures or conduits about 0.51 mm in diameter. The parts were supported on a vibrating table so that orifices, conduits and interstices, as small as 0.25 mm, were upwardly. Lower outlets of such orifices were taped to prevent egress of powder. Then, while the table vibrated, the orifices, conduits and interstices were filled with a powder of the following formulation:

Constituents	Parts by Weight
Co_2Al_9 #325 mesh	10.0
Chromium, #325 mesh	4.0
NH_4F, Ammonium Fluoride	0.75
Calcined aluminum oxide, #100 mesh	85.25

This is designated as the RB505-C blend for applications requiring high hot corrosion resistance of internal surfaces.

After the interstices were filled and the upper outlets taped shut, vibrating was continued for about two minutes. Thereupon, the turbine blades were carefully packed in

powder of the following formula, the RB505-B blend.

Constituents	Parts by Weight
Co_2Al_9 #325 mesh	3.0
Chromium #325 mesh	4.0
NaF	0.5
Calcined Aluminium oxide #100 mesh	92.5

Thereupon, the heat treating step was carried out at about 1052°C for two hours in an argon atmosphere and an excellent codeposited diffusion coating was obtained simultaneously on the interior and exterior surfaces of the articles being treated.

It is also to be understood that the following claims are intended to cover all of the generic and specific features of the invention herein described and all statements of the scope of the invention which might be said to fall therebetween.

Claims:

1. In a method of codeposition of aluminum and chromium diffusion coatings on articles formed of nickel-based superalloys, said method comprising the step of thermally treating said articles, while said articles are packed in a coating powder from which powder aluminum and chromium may be derived and which aluminum and chromium helps form a protective coating on the articles being treated, characterised by comprising the step of adding an effective quantity of the compound Co_2Al_9 into said coating.

2. A process as defined in Claim 1, characterised in that said article being treated is formed of a superalloy which comprises substantially 16% or less of chromium.

3. A process as defined in Claim 1, characterised in that said process is carried out on a superalloy comprising a maximum of substantially 14% chromium and within substantially two hours at a temperature of substantially 1052°C.

4. A process as defined in Claim 1, characterised in that the article to be processed contains internal surfaces and wherein the process comprises the steps of simultaneously treating said internal surfaces with a second powder pack composition comprising at least twice as much Co_2Al_9 as does a first powder pack composition which is used to treat the external surfaces of said article.

5. An improved process for treating the internal surfaces of articles formed of nickel-based and cobalt-based

superalloys, with a diffusion coating of chromium and aluminum codeposed at elevated temperatures from a powder composition in which said article is packed, characterised by

said method comprising the steps of

(1) using Co_2Al_9 as an aluminum-contributing chemical in said powder pack.

(2) using Cr (passes 325 mesh) powder as the chromium contributing chemical in said powder pack.

6. A particulate composition of matter forming means from which to expel aluminum and chromium at elevated temperatures,
characterised by
said composition comprising an inert filler, effective quantities of a halide carrier, chromium metal powder, and an aluminum-contributing powder of the compound Co_2Al_9.

7. A composition as defined in Claim 6,
characterised in that
said halide carrier is NaF.

8. A composition as defined in Claim 6 or 7,
characterised in that
the composition comprises from 1 to 15 percent of Co_2Al_9.

9. A composition as defined in Claims 6, 7 or 8,
characterised in that
the composition comprises from 2 percent to 6 percent of chromium.

10. A composition as defined in Claims 6, 7, 8 or 9,
characterised in that
said powder is resistant to being immobilised by a heat

treatment at substantially 1093°C and remains readily flowable after two hours of such a treatment.

11. A composition as defined in Claim 8, characterised in that the composition comprises from 2 percent to 6 percent of chromium.

12. A composition as defined in Claim 6, characterised in that said halide carrier is NH_4F.

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EUROPEAN SEARCH REPORT

0066019
Application number
EP 81302453.6

DOCUMENTS CONSIDERED TO BE RELEVANT			CLASSIFICATION OF THE APPLICATION (Int. Cl.³)
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	
X	GB - A - 1 315 229 (UNITED AIRCRAFT CORPORATION) * Example; claims * ----	1-3,6, 9-11	C 23 C 9/02

TECHNICAL FIELDS SEARCHED (Int. Cl.³)

C 23 C

CATEGORY OF CITED DOCUMENTS

X: particularly relevant
A: technological background
O: non-written disclosure
P: intermediate document
T: theory or principle underlying the invention
E: conflicting application
D: document cited in the application
L: citation for other reasons

&: member of the same patent family, corresponding document

X The present search report has been drawn up for all claims

Place of search	Date of completion of the search	Examiner
VIENNA	12-01-1982	SLAMA

EPO Form 1503.1 06.78