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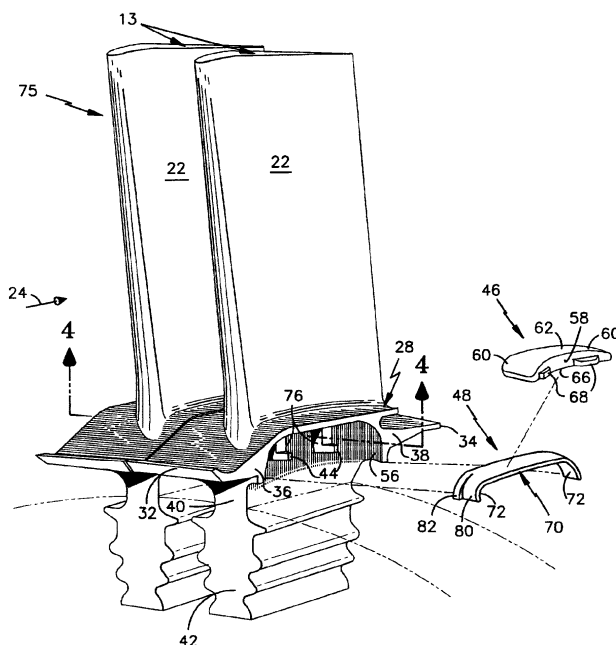
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(54) **Turbine blade platform seal**

(57) A seal for a turbine blade in a gas turbine engine has a sealing portion 72 with two subportions 80,82, where the subportions 80,82 are longitudinally offset

from one another, so that the seal may provide sealing for adjacent turbine blades having longitudinally offset inner platform surfaces.

FIG.3



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			F01D
The present search report has been drawn up for all claims			
Place of search MUNICH		Date of completion of the search 28 February 2000	Examiner Acton, P
CATEGORY OF CITED DOCUMENTS X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document			

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