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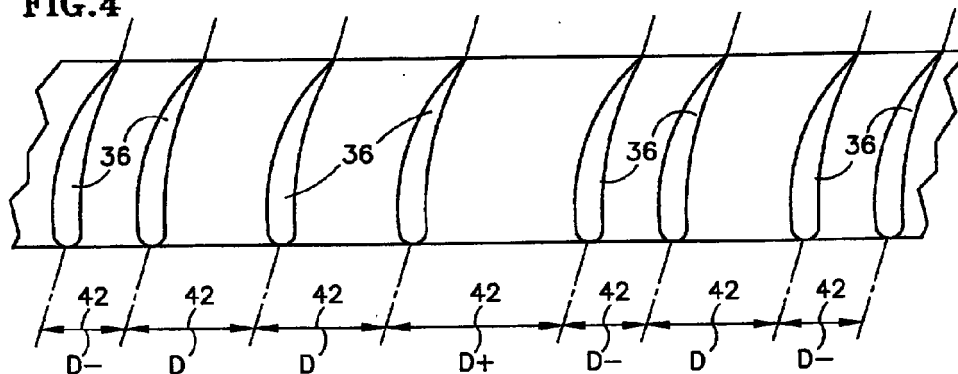
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(54) **Aerodynamically damping vibrations in a rotor stage of a turbomachine**

(57) A rotor stage for a gas turbine engine is provided which includes a rotor disk (34) and a plurality of rotor blades (36). The rotor blades extend radially outward from the outer radial surface of the disk, and are

separated from one another by a distance (D). The distance is selectively varied to increase aerodynamic damping of the rotor blades.

FIG.4





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EUROPEAN SEARCH REPORT

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CATEGORY OF CITED DOCUMENTS X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document			

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