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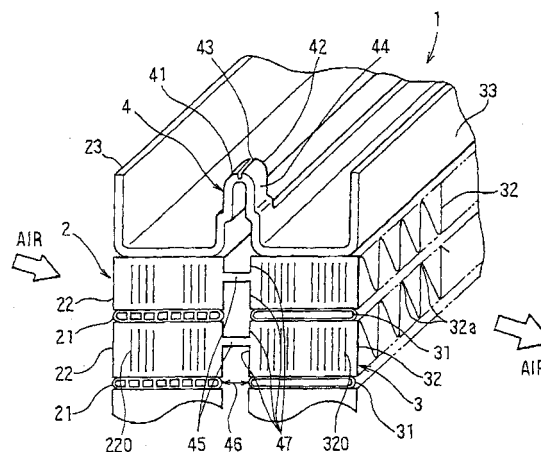
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(54) **Heat exchanger having several heat exchanging portions**

(57) A ratio ( $N_c/L_c$ ), in a condenser core portion (2), of the number ( $N_c$ ) of louvers (220) to a width ( $L_c$ ) of a condenser cooling fin (22), and a ratio ( $N_r/L_r$ ), in a radiator core portion (3), of the number ( $N_r$ ) of louvers to a width ( $L_r$ ) of a radiator cooling fin (32) satisfy that the ratio in one core portion, out of the condenser and the radiator core portions, a required radiation amount of which is larger than that of the other core portion is larger than the ratio in the other core portion.

Thus, in the core portion having a small required radiation amount, the number of louvers relative to the width of the cooling fin is small thereby decreasing the heat transfer ratio. However, by this, the air flow resistance in this core portion decreases thereby increasing an air flow amount. Thus, the radiation amount of the core portion having a large required radiation amount increases.

**FIG. 1**



**Description****BACKGROUND OF THE INVENTION**

## 1. Field of the Invention

**[0001]** The present invention relates to a heat exchanger in which different core portions are integrated with each other, and more particularly the present invention relates to a heat exchanger which can be effectively applied to a radiator of an automotive engine and a condenser of an automotive air conditioning apparatus.

## 2. Description of Related Art

**[0002]** Conventionally, an automotive air conditioning apparatus is assembled into a vehicle at a car dealer or the like after the vehicle has been completed. Recently, however, the automotive air conditioning apparatus is generally installed in the vehicle during vehicle assembling process. Therefore the automotive air conditioning apparatus is assembled with automotive parts in the assembling process of the vehicle at the manufacturing plant.

**[0003]** A heat exchanger in which different core portions such as a radiator and a condenser are integrated is disclosed in Japanese Patent Publication No. 3-177795. In this heat exchanger, cooling fins of first core portion and second core portion are integrated with each other. These cooling fins are connected to each oval flat tube of the first and second core portions by brazing.

**[0004]** In the cooling fin, a plurality of slits are formed at the center portion between the first and second core portions for interrupting a heat transmission from a high temperature side core portion (for example, radiator core portion) to a low temperature side core portion (for example, condenser core portion).

**[0005]** The required heat exchanging abilities of the first core portion (condenser core portion) and the second core portion (radiator core portion) varies in accordance with the difference of engine type or vehicle type despite the required constitutions of the heat exchanger are the same. When the automotive heat exchanger is constructed by some single heat exchangers, the required heat exchanging abilities thereof are set by tuning fin pitches of the cooling fins respectively in accordance with the engine type or vehicle type.

**[0006]** However, in the heat exchanger in which different core portions are integrated and cooling fins of first core portion and second core portion are integrated with each other, each fin pitch cannot be designed independently respectively. Therefore, the above-described method of setting the fin pitches in the first and second core portions respectively cannot be applied to this type heat exchanger.

**SUMMARY OF THE INVENTION**

**[0007]** In view of the foregoing problems, it is an object of the present invention to provide a heat exchanger in which different core portions and cooling fins thereof are integrated with each other, while setting the required heat exchanging abilities of each core portion independently respectively.

**[0008]** According to a first aspect of the present invention, a ratio, in a first core portion, of the number of louvers to a width of a first cooling fin, and a ratio, in a second core portion, of the number of louvers to a width of a second cooling fin are set to be in such a manner that the ratio in one core portion, out of said first and second core portion, the required radiation amount of which is larger than that of the other core portion is larger than the ratio in the other core portion.

**[0009]** Thus, in the core portion having a small required radiation amount, the number of louvers relative to the width of the cooling fin is small thereby decreasing the heat transfer ratio. However, the pressure loss in this core portion decreases thereby increasing the amount of an external fluid. Thus, the radiation amount of the core portion having a large required radiation amount increases.

**[0010]** According to a second aspect of the present invention, in one core portion, out of the first and second core portions, the required radiation amount of which is smaller than that of the other core portion, a width of the cooling fin in an external fluid flow direction is shorter than a width of a tube in its cross sectionally longitudinal direction. Further, a ratio, in the first core portion, of the number of louvers to the width of a first tube, and a ratio, in the second core portion, of the number louvers to the width of a second tube are set to be in such a manner that the ratio in one core portion, out of the first and second core portions, the required radiation amount of which is smaller than that of the other core portion is smaller than the ratio in the other core portion.

**[0011]** Thus, in the core portion having a small required radiation amount, the width of the cooling fin and the number of louvers relative to the width of the tube in its cross sectionally longitudinal direction are small thereby decreasing the heat transfer ratio. However, by this, the pressure loss in the core portion decreases thereby increasing the amount of an external fluid. Thus, the radiation amount of the core portion having a large required radiation amount increases.

**[0012]** According to a third aspect of the present invention, the length of the louver in one core portion, out of the first and second core portions, the required radiation amount of which is smaller than that of the other core portion is shorter than the length of the louver in the other core portion.

**[0013]** Thus, in the core portion having a small required radiation amount, the length of the louver is short thereby decreasing the heat transfer ratio. However, by this, the pressure loss in the core portion decreases thereby increasing the flow amount of the external fluid. Thus, the radiation amount of the core portion having a large required radiation amount increases.

**[0014]** According to a fourth aspect of the present invention, a tilt angle of the louver in one core portion, out of the first and second core portion, the required radiation amount of which is smaller than that of the other core portion is smaller than the tilt angle of the louver in the other core portion.

**[0015]** Thus, in the core portion having a small required radiation amount, the tilt angle of the louver is small thereby decreasing the heat transfer ratio. However, by this, the pressure loss in the core portion decreases thereby increasing the flow amount of the external fluid. Thus, the radiation amount of the core portion having a large required radiation amount increases.

## BRIEF DESCRIPTION OF THE DRAWINGS

**[0016]** Additional objects and advantages of the present invention will be more readily apparent from the following detailed description of preferred embodiments thereof when taken together with the accompanying drawings in which;

FIG. 1 is a perspective view showing a core portion of a heat exchanger according to the first embodiment of the present invention;

FIG. 2 is a front view showing a core portion of a heat exchanger according to the first embodiment;

FIG. 3 is a plan view showing a core portion of a heat exchanger according to the first embodiment;

FIG. 4 is a perspective view showing a shape of the cooling fin;

graph showing a relationship between an increase percentage of radiating amount of a cooling fin in a condenser and a projection length of the cooling fin;

FIG. 5A is a plan view showing tubes and cooling fins according to the first embodiment, FIG. 5B is a cross sectional view taken along line 5B-5B in FIG. 5A;

FIG. 6A is a plan view showing tubes and cooling fins according to the second embodiment, FIG. 6B is a cross sectional view taken along line 6B-6B in FIG. 6A;

FIG. 7 is a graph showing a relationship between a number of louvers decreasing ratio and a performance ratio;

FIG. 8A is a plan view showing tubes and cooling fins according to the third embodiment, FIG. 8B is a cross sectional view taken along line 8B-8B in FIG. 8A;

FIG. 9A is a plan view showing tubes and cooling fins according to the fourth embodiment, FIG. 9B is a cross sectional view taken along line 9B-9B in FIG. 9A;

FIG. 10A is a plan view showing tubes and cooling fins according to the fourth embodiment, FIG. 10B is a cross sectional view taken along line 10B-10B in FIG. 10A;

FIG. 11A is a plan view showing tubes and cooling fins according to the sixth embodiment, FIG. 11B is a cross sectional view taken along line 11B-11B in FIG. 11A;

FIG. 12 is a graph showing a relationship between a fin width ratio and a performance ratio;

FIG. 13A is a plan view showing tubes and cooling fins according to the seventh embodiment, FIG. 13B is a cross sectional view taken along line 13B-13B in FIG. 13A;

FIG. 14A is a plan view showing tubes and cooling fins according to the first comparison example of the seventh embodiment, FIG. 14B is a cross sectional view taken along line 14B-14B in FIG. 14A;

FIG. 15A is a plan view showing tubes and cooling fins according to the second comparison example of the seventh embodiment, FIG. 15B is a cross sectional view taken along line 15B-15B in FIG. 15A;

FIG. 16 is a graph showing the relations between a number of louvers and a performance ratio;

FIG. 17 is a graph showing a flat turning portion length and a performance ratio;

FIG. 18 is a graph showing a heat transfer ratio in accordance with a position of the cooling fin along an air flow direction;

FIG. 19A is a plan view showing tubes and cooling fins according to the eighth embodiment, FIG. 19B is a cross sectional view taken along line 19B-19B in FIG. 19A;

FIG. 20A is a plan view showing tubes and cooling fins according to the ninth embodiment, FIG. 20B is a cross sectional view taken along line 20B-20B in FIG. 20A;

FIG. 21A is a plan view showing tubes and cooling fins according to the tenth embodiment, FIG. 21B is a cross sectional view taken along line 21B-12B in FIG. 21A;

FIG. 22 is a graph showing relations between a louver cut length ratio and a performance ratio;

FIG. 23A is a plan view showing tubes and cooling fins according to the eleventh embodiment, FIG. 23B is a cross sectional view taken along line 23B-23B in FIG. 23A;

FIG. 24A is a plan view showing tubes and cooling fins according to the twelfth embodiment, FIG. 24B is a cross sectional view taken along line 24B-24B in FIG. 24A;

FIG. 25A is a plan view showing tubes and cooling fins according to the thirteenth embodiment, FIG. 25B is a cross sectional view taken along line 25B-25B in FIG. 25A; and

FIG. 26 is a graph showing relations between louver a tilt angle reduction ratio and a performance ratio.

#### DETAILED DESCRIPTION OF PRESENTLY PREFERRED EXEMPLARY EMBODIMENTS

**[0017]** Preferred embodiments of the present invention are described hereinafter with reference to the accompanying drawings.

(First Embodiment)

**[0018]** In an automotive heat exchanger 1 shown in FIGS. 1,2, a condenser core portion 2 of an automotive air conditioning apparatus is used as a first core portion, and a radiator core portion 3 for cooling an engine is used as a second core portion. Generally, because the temperature of refrigerant flowing through the condenser core portion 2 is lower than that of engine cooling water flowing through the radiator core portion 3, the condenser core portion 2 is disposed at the upstream air side of the radiator core portion 3 in air flow direction and the two core portions 2, 3 are disposed in series in the air flow direction at the front-most portion of an engine compartment. The structure of the heat exchanger of the first embodiment is hereinafter described with reference to FIGS. 1 through 5.

**[0019]** FIG. 1 is a partial enlarged cross-sectional view of a heat exchanger 1 of the present invention. As shown in FIG. 1, a condenser core portion 2 and a radiator core portion 3 are disposed in series in the air flow direction so as to form predetermined clearances 46 between each pair of a condenser tube 21 and a radiator tube 31 described later to interrupt heat transmission.

**[0020]** The condenser core portion 2 includes flat shaped condenser tubes 21 in which a plural refrigerant passages are formed, and corrugated (wave-shaped) cooling fins 22 in which a plurality of folded portions 22a brazed to the condenser tube 21 are formed.

**[0021]** The radiator core portion 3 has a similar structure with the condenser core portion 2. The radiator core portion 3 includes the radiator tubes 31, in which a single refrigerant passage is formed, disposed in parallel with the condenser tubes 21 and radiator cooling fins 32. The tubes 21 and 31 and the cooling fins 22, 32 are alternately laminated and are brazed to each other. A plurality of louvers 220 and 320 are formed in the two cooling fins 22, 32 to facilitate heat exchange. The two cooling fins 22, 23 and a plurality of connecting portions 45 are integrally formed with the louvers 220, 320 by a roller forming method or the like.

**[0022]** The connecting portions 45 are formed between the two cooling fins 22, 32 for connecting the two cooling fins 22, 23. At both sides of the connecting portion 45, adiabatic slits 47 are provided for interrupting heat transmission from the radiator core portion 3 to the condenser core portion 2. The width of the connecting portion 45 is set to be smaller enough than the height of the cooling fins 22, 32 (the distance between a pair of adjacent flat tubes 21, 31) to suppress the heat transmission from the radiator core portion 3 to the condenser core portion 2.

**[0023]** Side plates 23, 33 are reinforcement member of the two heat exchanging core portions 2, 3. The side plates 23, 33 are respectively disposed in upper and lower end portions of the two heat exchanging core portions 2, 3 as shown in FIG. 2. As shown in FIG. 1, the side plates 23, 33 are integrally formed from a sheet of aluminum plate to a general U-shape in cross section. Connecting portions 4 for connecting the side plate 23 and the side plate 33 are formed in two end portions of the longitudinal direction of the two side plates 23, 33. A Z-shaped bent portion 41 of the side plate 23 and a Z-shaped bent portion 42 of the side plate 33 are connected to each other at a top end portion 43 so that the connecting portion 4 is formed. The width of the connecting portion 4 is set to be small enough as compared with the dimension of the side plate 23 or 33 in the longitudinal direction to suppress the heat transmission. Further, a recess portion is formed in the top end portion 43 of the connecting portion 4 to reduce the thickness of the plate wall of the connecting portion 4.

**[0024]** Further, as shown in FIG. 2, a first header tank 34 for distributing cooling water to each radiator tube 31 is disposed at an end (left end) side of the radiator core portion 3. The front shape of first header tank 34 is nearly a triangular, the cross-sectional shape is ellipsoid as shown in FIG. 3. An inlet 35 of cooling water flowing to the radiator is formed at an upper side of the first header tank 34 having a nearly triangular shape. Further, a pipe 35a for connecting a pipe (not shown) of cooling water is brazed to the inlet 35.

**[0025]** Further, a second header tank 36 for receiving the cooling water having been heat-exchanged is disposed in an opposite end (right end) of the first header tank 34. The second header tank 36 has a similar shape with the first header tank 34. As shown in FIG. 2, the second header tank 36 and the first header tank 34 are point-symmetrical with

reference to the center of the radiator core portion 3. Further, an outlet 37 for discharging the cooling water is formed at the bottom side of the second header tank 36. with the tubes and the cooling fins and the like, a pipe 37a for connecting the pipe (not shown) of cooling water is brazed to the outlet 37.

**[0026]** A first header tank 24 is disposed at an end side of the condenser core portion 2 for distributing the refrigerant into each condenser tube 21, and the body of the first header tank 24 is cylindrically formed as shown in FIG. 3. The first header tank 24 of the condenser is disposed to have a predetermined clearance with the second header tank 36 of the radiator. Further, a joint 26a for connecting a refrigerant pipe (not shown) is brazed to the body of the first header tank 24, and an inlet 26 of refrigerant is formed in the joint 26a.

**[0027]** Further, as shown in FIG. 3, a second header tank 25 of the condenser for receiving the refrigerant having been heat-exchanged is disposed at an opposite end of the first header tank 24 of the condenser core portion 2. The second header tank 25 is disposed to have a predetermined clearance with the first header tank 34 of the radiator. The body of the second header tank 25 is cylindrically formed. Further, as shown in FIG. 2, a joint 27a for connecting a refrigerant pipe (not shown) is brazed to the body of the second header tank 25. An outlet 27 of refrigerant is formed in the joint 27a.

**[0028]** Next, the condenser cooling fin 22 and the radiator cooling fin 32 will be described.

**[0029]** The width  $L_c$  of the condenser cooling fin 22 and the width  $L_r$  of the radiator cooling fin 32 have the same length as the width of the tubes 21, 31 in the cross sectional longitudinal direction thereof. Here, the widths  $L_c$ ,  $L_r$  are the dimension of the cooling fins 22, 32 along the cross sectionally longitudinal direction of the tubes 21, 31 (air flow direction).

**[0030]** The louver 220 of the condenser cooling fin 22 is constructed by a first louver group 221, a second louver group 222, and a turning louver 223 arranged between both louver groups 221, 222. The turning louver 223 turns the air flow. The first louver group 221 and the second louver group 222 tilt toward the opposite side to each other.

**[0031]** Similarly, a first louver group 321, a second louver group 322, and a turning louver 323 are provided in the radiator cooling fin 32.

**[0032]** The numbers of both louvers 220, 320 are set as follows to improve the heat transmitting ability (heat transmitting amount). In the condenser cooling fin 22, each first and second louver groups 221, 222 has three louvers 220. In the radiator cooling fin 32, each first and second louver groups 321, 322 has five louvers 320.

**[0033]** That is, the number  $N_c$  of the louvers 220 in the condenser cooling fin 22 is six ( $N_c=6$ ), and the number  $N_r$  of the louvers 320 in the radiator cooling fin 32 is ten ( $N_r=10$ ).

**[0034]** Accordingly, the ratio of the  $N_c$  and  $L_c$  in the condenser cooling fin 22 ( $N_c/L_c$ ) and the ratio of the  $N_r$  and  $L_r$  in the radiator cooling fin 32 ( $N_r/L_r$ ) satisfy the following relation:

$$(N_c/L_c) < (N_r/L_r).$$

**[0035]** Here, the condenser cooling fin 22 has six louvers although ten louvers can be provided thereon if desired. Therefore, the area of air introducing portions 224, 225 provided in front and rear of the louvers 220 can be wide relative to the area where the louvers 220 are formed.

**[0036]** Accordingly, the ratio of the sum of the lengths of the air introducing portions 224, 225 in the air flow direction ( $L_1+L_2$ ) to the length of the space where the louvers 220 are formed in the air flow direction  $L_3$ ,  $[(L_1+L_2)/L_3]$ , and the ratio of the sum of the lengths of the air introducing portions 324, 325 in the air flow direction ( $L_4+L_5$ ) to the length of the space where the louvers 320 are formed in the air flow direction  $L_6$ ,  $[(L_4+L_5)/L_6]$ , satisfy the following relation:

$$[(L_1+L_2)/L_3] > [(L_4+L_5)/L_6].$$

**[0037]** Next, an operation of the above-described structure will be explained.

**[0038]** When a cooling fan (not illustrated) which is disposed at the air downstream side of the radiator core portion 3 operates, the cooling air passes through the condenser core portion 2 and the radiator core portion 3, as shown in FIGS. 1 and 2.

**[0039]** At the same time, a gas phase refrigerant flowing out of a compressor flows into the first header tank 24 through the refrigerant inlet 26. The gas phase refrigerant flows in the condenser tubes 21 from the right side to the left side in FIGS. 2 and 3 while being heat exchanged with the cooling air to be condensed. The condensed liquid phase refrigerant is collected in the second header tank 25 and flows out of the condenser core portion 2 through the refrigerant outlet 27.

**[0040]** A hot engine coolant flows from an engine into the first header tank 34 through the engine coolant inlet 35. The engine coolant flows in the radiator tube 31 from the left side to the right side in FIGS. 2 and 3 while being heat exchanged with the cooling air to be cooled. The cooled engine coolant is collected in the second header tank 36 and

flows out of the radiator core portion 3 through the engine coolant outlet 37.

**[0041]** The heat exchanging abilities of the condenser core portion 2 and the radiator core portion 3, if the constitutions thereof are the same, depend on the heat transmitting ratio and the air flow resistance thereof. The heat transmitting ratio and the air flow resistance decrease in accordance with a decrease in the number of the louvers 220, 320.

**[0042]** According to the first embodiment, in the condenser cooling fin 22, six louvers are provided although ten louvers can be provided thereon if desired. While, in the radiator cooling fin 32, ten louvers are provided by using the most of the space thereof.

**[0043]** Therefore, the heat transfer ratio in the condenser core portion 2 decreases in accordance with the decreasing the number of the louvers 220. Thus, the heat transmitting ability of the condenser core portion 2 decreases. However, the air flow resistance in the condenser core portion 2 decreases thereby increasing the amount of the cooling air passing through the radiator core portion 3. Thus, the heat transmitting ability of the radiator core portion 3 increases.

(Second Embodiment)

**[0044]** According to the second embodiment, as shown in FIGS. 6A, 6B, in the condenser cooling fin 22, ten louvers 220 are provided by making the most of the space thereof. While, in the radiator cooling fin 32, six louvers 320 are provided although ten louvers can be provided thereon if desired. That is, the relation:  $(N_c/L_c) > (N_r/L_r)$  is satisfied. Thereby, the radiation amount in the radiator core portion 3 decreases, while the radiation amount in the condenser core portion 2 increases with the air flow amount increasing.

**[0045]** FIG. 7 shows the relations between the number of louvers decreasing ratio and the performance ratios or the core portions 2, 3 under the condition that air flow speed of the cooling air is constant. Here, the number of louvers decreasing ratio is defined as a ratio of the number of louvers decreased relative to the number of louvers which can be provided within the predetermined fin width  $L_c$ ,  $L_r$ . For example, in the condenser cooling fin 22 shown in FIG. 5A, six louvers is provided although ten louvers can be provided, thus the number of louvers decreasing ratio is 40%. Similarly, in the radiator cooling fin 32 shown in FIG. 6A, the number of louvers decreasing ratio is 40%.

**[0046]** As is understood from FIG. 7, when the number of louvers decreasing ratio is set to 50% in one of the condenser core portion 2 and the radiator core portion 3, the radiation amount in this core portion decreases by about 10% and the pressure loss therein decreases by about 30%. In this way, as the pressure loss decreases in one core portion, the flow amount of the air passing through these core portions increases thereby increasing the radiation amount in the other core portion by about 5%.

**[0047]** Further, as is understood from FIG. 7, it is necessary to set the number of louvers decreasing ratio to 30% or more for decreasing the pressure loss by about 20%.

(Third Embodiment)

**[0048]** According to the third embodiment, as shown in FIGS. 8A, 8B, a projection portion 326 is formed at the air upstream side end (the end facing the condenser core portion 2) of the radiator cooling fin 32. This projection portion 326 protrudes from the end of the radiator tube 31 toward the air upstream side. Thereby, the number of louvers  $N_r$  in the radiator cooling fin 32 is increased more than that in the first embodiment.

**[0049]** For example, as shown in FIGS. 8A, 8B, the radiator cooling fin 32 has twelve louvers 320. Thus, a radiation amount difference between in the condenser core portion 2 and in the radiator core portion 3 is expanded more than in the first embodiment.

(Fourth Embodiment)

**[0050]** According to the fourth embodiment, as described in the first embodiment, the condenser cooling fin 22 has six louvers in spite of ten louvers can be provided thereon if making the most of the space thereof. In the fourth embodiment, as shown in FIGS. 9A, 9B, the louver pitch  $L_{pc}$  of the louver 220 is set to be wider than the louver pitch  $L_{pr}$  of the louver 320. Here, the louver pitch  $L_{pc}$  is defined as a distance between a pair of adjacent louvers 220, 320. This distance is same as the length of each louver 220, 320 in the air flow direction.

**[0051]** In this way, the louver pitch in the condenser cooling fin 22 is set to be wider than in the first embodiment. Thus, the length of the air introducing portions 224, 225 ( $L_1+L_2$ ) can be decreased more than in the first embodiment.

**[0052]** In the first embodiment, the area  $L_3$  where the louvers 220 are formed is partial to the center portion of the condenser cooling fin 22. Thus, the air flowing along the tilted surface of the louvers 220 is collected in the center portion of the cooling fin 22, and the reduction ratio of the heat transmitting ratio can be made remarkable. However, in the fourth embodiment, as the louver pitch  $L_{pc}$  is set to be larger than in the first embodiment, the air flowing along the tilted surface of the louvers 220 is spread entirely. Thus, the reduction ratio of the heat transmitting ratio can be decreased.

(Fifth Embodiment)

**[0053]** According to the fifth embodiment, as shown in FIGS. 10A, 10B, the fin width  $L_c$  of the condenser cooling fin 22 is smaller than the width  $L_{tc}$  of the condenser oval flat tube 21. While, in the radiator cooling fin 32, the fin width  $L_r$  is same as the width  $L_{tr}$  of the radiator oval flat tube 31. Here, the width  $L_{tc}$  of the condenser tube 21 is same as the width  $L_{tr}$  of the radiator tube 31.

**[0054]** Accordingly, the ratio of the number of louvers 220  $N_c$  (in FIGS. 10A, 10B,  $N_c=6$ ) to the condenser tube width  $L_{tc}$  ( $N_c/L_{tc}$ ) and the ratio of the number of louvers 320  $N_r$  (in FIGS. 10A, 10B,  $N_r=10$ ) to the radiator tube width  $L_{tr}$  ( $N_r/L_{tr}$ ) satisfy the following relation:

$$(N_c/L_{tc}) < (N_r/L_{tr}).$$

**[0055]** Here, in FIGS. 10A, 10B,  $L_f$  denotes a width of an entire fin constructed by the condenser cooling fin 22 and the radiator cooling fin 32, and  $L$  denotes the distance between both ends of both oval flat tubes 21, 31 (the width of the heat exchanger).

**[0056]** According to the fifth embodiment, because in the condenser core portion 2, the fin width  $L_c$  relative to the tube width  $L_{tc}$  is small in comparison with in the radiator core portion 3, the radiation area in the condenser core portion 2 decreases thereby decreasing the radiation amount. However, by decreasing the fin width  $L_c$  and the number  $N_c$  of the louvers 220 decreases, the air flow resistance in the condenser core portion 2 decreases thereby increasing the air flow amount passing through these heat exchanging core portions 2, 3. Consequently, the radiation amount in the radiator core portion 3 increases.

(Sixth Embodiment)

**[0057]** According to the sixth embodiment, as shown in FIGS. 11A, 11B, the fin width  $L_r$  of the radiator cooling fin 32 is smaller than the width  $L_{tr}$  of the radiator oval flat tube 31. While, in the condenser cooling fin 22, the fin width  $L_c$  is same as the width  $L_{tc}$  of the condenser oval flat tube 21. Here, the width  $L_{tc}$  of the condenser tube 21 is same as the width  $L_{tr}$  of the radiator tube 31.

**[0058]** Accordingly, the ratio of the number  $N_c$  of louvers 220 (in FIGS. 11A, 11B,  $N_c=10$ ) to the condenser tube width  $L_{tc}$  ( $N_c/L_{tc}$ ) and the ratio of the number  $N_r$  of louvers 320 (in FIG. 11A, 11B,  $N_r=6$ ) to the radiator tube width  $L_{tr}$  [ $N_r/L_{tr}$ ] satisfy the following relation:

$$(N_c/L_{tc}) > (N_r/L_{tr}).$$

**[0059]** Thus, the radiation amount in the radiator core portion 3 decreases. However, the air flow resistance in the radiator core portion 3 decreases thereby increasing the air flow amount passing through these heat exchanging core portions 2, 3. Consequently, the radiation amount in the condenser core portion 2 increases.

**[0060]** FIG. 12 is a graph showing the experimented results based on the fifth and the sixth embodiments. The graph shows relations between the ratio of the fin width  $L_c$ ,  $L_r$  to the tube width  $L_{tc}$ ,  $L_{tr}$  ( $L_c/L_{tc}$ ,  $L_r/L_{tr}$ ) and the radiation performance ratio of the condenser core portion 2 and the radiator core portion 3. Here, the experimented results are under the condition that the air flow speed is constant.

**[0061]** As is understood from FIG. 12, when the fin width  $L_c$  or  $L_r$  is set to 80% of the tube widths  $L_{tc}$ ,  $L_{tr}$  in one of the condenser core portion 2 and the radiator core portion 3, the radiation amount in this core portion decreases by about 10% and the pressure loss therein decreases by about 20%. In this way, as the pressure loss decreases in one core portion, the flow amount of the air passing through these core portions increases thereby increasing the radiation amount in the other core portion by about 3%. Further, as is understood from FIG. 12, it is necessary to set the fin width  $L_c$ ,  $L_r$  to 80% or less of the tube width  $L_{tc}$ ,  $L_{tr}$ .

(Seventh Embodiment)

**[0062]** According to the seventh embodiment, as shown in FIGS. 13A, 13B, the length  $L_T$  of the flat turning surface 223a, 323a of the turning louver 223, 323 is set to be three times or more as the louver pitch  $L_p$ . Here, for example, the length of the flat turning surface 223a, 323a is set to be about 5.5 times as the louver pitch  $L_p$ . The object of the seventh embodiment is to suppress the reduction of heat transfer ratio in the cooling fin 22, 32.

**[0063]** FIGS. 14 and 15 show a first and a second comparison examples being compared with the seventh embodiment. The first and second comparison examples are all the same except for the number of louvers 220, 320.

**[0064]** According to the experimented results and studies about the first and second comparison examples, when the number of louvers is simply decreased from both front and rear side in the air flow direction, both air pressure loss and heat transfer ratio are decreased proportionally, as shown in FIG. 16.

**[0065]** Further, according to the experimented results and studies about relations between the length  $L_T$  of the flat turning surface 223a, 323a of the turning louver 223, 323 and the performance ratio of the core portion 2, 3, when the length  $L_T$  of the flat turning surface 223a, 323a becomes large, both heat transfer ratio and pressure loss ratio of the fin increase as shown in FIG. 17. Here, FIG. 17 shows the relations between the length  $L_T$  and the performance ratio of the core portion 2, 3 under the condition that the air flow speed is constant. The length  $L_T$  is expressed as a multiple of the louver pitch  $L_p$ .

**[0066]** As is understood from FIG. 17, the heat transfer ratio and the pressure loss ratio of the fin increase as the length  $L_T$  becomes large, and are saturated as the length  $L_T$  is more than  $3 \times L_p$ . Therefore, it is preferable to set the length  $L_T$  to be three times or more as the louver pitch  $L_p$ .

**[0067]** The heat transfer ratio of the fin increases in accordance with that the length  $L_T$  of the flat turning surface 223a, 323a becomes large because the following reason. That is, as the length  $L_T$  becomes large, the flow speed of the air passing through the second louver group 222, 322 which is disposed at the air downstream side of the turning louver 223, 323 recovers. Thus, the air passes through the second louver group 222, 322 at high speed.

**[0068]** Accordingly, in the seventh embodiment, the length  $L_T$  of the flat turning surface 223a, 323a of the turning louver 223, 323 is set to be three times or more as the louver pitch  $L_p$ .

**[0069]** In FIG. 18A, the axis of abscissa denotes the cross sectional shape of the fin in the comparison example shown in FIG. 14B in the air flow direction. In FIG. 18B, the axis of abscissa denotes the cross sectional shape of the fin in the seventh embodiment shown in FIG. 13B in the air flow direction.

**[0070]** In the comparison example, the turning louver 223, 323 is formed into a V-shape, i.e., the turning louver 223, 323 has no flat turning surface. Thus, the flow speed of the air passing through the second louver group 222, 322 does not recover and is still low. Therefore, as denoted by ① in FIG. 18A, the heat transfer ratio in the second louver group 222, 322 is lower than that in the first louver group 221, 321.

**[0071]** Contrary to this, in the seventh embodiment, the length  $L_T$  of the flat turning surface 223a, 323a is set to be 5.5 times as the louver pitch  $L_p$ . That is, the length  $L_T$  is large enough to make the speed of the air passing through the second louver group 222, 322 recover. Thus, because the air passes through the second louver group 222, 322 at high speed, the heat transfer ratio in the second louver group 222, 322 is approximately the same as in the first louver group 221, 321 as denoted by ② in FIG. 18B.

**[0072]** According to the inventor's research and study, it is preferable that the length  $L_T$  of the flat turning surface 223a, 323a in one cooling fin in which the number of louvers is smaller than that in the other cooling fin is set to be longer than the length  $L_i$  of the air introducing portion 224, 324 disposed at the air upstream side of the louvers 220, 320 for making the flow speed of the air passing through the second louver group 222, 322 recover.

(Eighth Embodiment)

**[0073]** According to the eighth embodiment, as shown in FIGS. 19A, 19B, a length (cut length)  $E_c$  of the condenser louver 220 and a length (cut length)  $E_r$  of the radiator louver 320 are set to be different from each other. The length  $E_c$ ,  $E_r$  is defined as a length of the louver 220, 320 in a direction perpendicular to the air flow direction, and influences the heat transfer ratio and the air flow resistance.

**[0074]** That is, when the length  $E_c$ ,  $E_r$  of the louver 220, 320 is decreased, the heat transfer ratio and the air flow resistance are also decreased.

**[0075]** In the eighth embodiment, the length  $E_c$  of the condenser louver 220 is set to be shorter than the length  $E_r$  of the radiator louver 320 for improving the performance of the radiator core portion 3.

**[0076]** Thus, though the performance of the condenser core portion 2 is decreased by shortening the length  $E_c$  of the condenser louver 220, the air resistance is decreased by shortening the length  $E_c$  of the condenser louver 220 thereby increasing the air flow amount. Therefore, the performance of the radiator core portion 3 is improved.

**[0077]** Here, for example, the fin height  $H_f$  of the cooling fin 22, 32 (distance between a pair of adjacent tubes) is 8mm, the length  $E_r$  of the radiator louver 320 is 7mm, and the length  $E_c$  of the condenser louver 220 is 5mm.

(Ninth Embodiment)

**[0078]** According to the ninth embodiment, as shown in FIGS. 20A, 20B, the length  $E_r$  of the radiator louver 320 is set to be shorter than the length  $E_c$  of the condenser louver 220 for improving the performance of the condenser core portion 2.



(Tenth Embodiment)

**[0079]** According to the tenth embodiment, as shown in FIGS. 21A, 21B, the projection portion 326 described in FIG. 8A is provided at the air upstream side end of the radiator cooling fin 32, and a projection portion 327 facing the projection portion 326 is provided at the air downstream side end of the condenser cooling fin 22 also. By this, the number of condenser louvers 220 in the second louver group 222 and the number of radiator louvers 320 in the first louver group 321 are increased.

**[0080]** Further, the length  $E_c$  of the condenser louver 220 is set to be shorter than the length  $E_r$  of the radiator louver 320.

**[0081]** FIG. 22 is a graph showing relations between the length of the louver in the eighth through tenth embodiments and the performance of the core portion under the condition that the flow speed of the air passing through the core portion is constant. The louver length ratio placed on the axis of abscissa is a ratio of the louver length which is shortened intently (for example, condenser louver length  $E_c$  in the eighth embodiment) to the louver length which is defined by the fin height  $H_f$  (for example, radiator louver length  $E_r$  in the eighth embodiment),

**[0082]** That is, the louver length ratio is defined as follows:

$$(\text{Louver length which is shortened intently})/(\text{Louver length which is defined by a fin height}).$$

**[0083]** As is understood from FIG. 22, when the louver length ratio is set to be 50%, the radiation amount in the core portion in which the louver length is shorten decreases by about 10%, and the pressure loss therein decreases by about 30%. By this, pressure loss decreases by about 30%, the radiation amount in the core portion in which the louver length is defined by the fin height is improved by about 5%.

(Eleventh Embodiment)

**[0084]** According to the eleventh embodiment, as shown in FIGS. 23A, 23B, a tilt angle  $\theta_c$  of the condenser louver 220 and a tilt angle  $\theta_r$  of the radiator louver 320 are set to be different from each other. The tilt angles  $\theta_c$ ,  $\theta_r$  influence the heat transfer ratio and the air flow resistance.

**[0085]** That is, when the tilt angle  $\theta_c$ ,  $\theta_r$  of the louver 220, 320 is decreased, the speed of the air passing through the louvers is decreased, and the heat transfer ratio and the air flow resistance are also decreased.

**[0086]** In the eleventh embodiment, the tilt angle  $\theta_c$  of the condenser louver 220 is set to be smaller than the tilt angle  $\theta_r$  of the radiator louver 320 for improving the radiation performance of the radiator core portion 3,

**[0087]** Thus, though the performance of the condenser core portion 2 decreases by reducing the tilt angle  $\theta_c$  of the condenser louver 220, the air resistance decreases by reducing the tilt angle  $\theta_c$  of the condenser louver 220 thereby increasing the air flow amount. Therefore, the performance of the radiator core portion 3 is improved.

**[0088]** For example, the tilt angle  $\theta_c$  of the condenser louver 220 is  $18^\circ$ , and the tilt angle  $\theta_r$  of the radiator louver 320 is  $25^\circ$ .

(Twelfth Embodiment)

**[0089]** According to the twelfth embodiment, as shown in FIGS. 24A, 24B, the tilt angle  $\theta_r$  of the radiator louver 320 is set to be smaller than the tilt angle  $\theta_c$  of the condenser louver 220 for improving the performance of the condenser core portion 2.

(Thirteenth Embodiment)

**[0090]** According to the thirteenth embodiment, as shown in FIGS. 25A, 25B, the projection portion 326 described in FIG. 21 is provided at the air upstream side end of the radiator cooling fin 32, and a projection portion 327 facing the projection portion 326 is provided at the air downstream side end of condenser cooling fin 22 also. By this, the number of condenser louvers 220 in the second louver group 222 and the number of radiator louvers 321 in the first louver group 322 are increased.

**[0091]** Further, the tilt angle  $\theta_c$  of the condenser louver 220 is set to be larger than the tilt angle  $\theta_r$  of the radiator louver 320.

**[0092]** FIG. 26 is a graph showing relations between the tilted angle of the louver in the eleventh through thirteenth embodiments and the performance of the core portion under the condition that the flow speed of the air passing through the core portion is constant.

**[0093]** Here, a louver tilt angle reduction ratio which is placed on the axis of abscissa is defined as a ratio of the tile-

angle reduced intently to the common tilt-angle for attaining a high heat transfer ratio.

**[0094]** That is, the louver tilt angle reduction ratio is defined as follows:

$$(\text{tile-angle reduced intently})/(\text{common tilt-angle for attaining a high heat transfer ratio}) \times 100.$$

**[0095]** As is understood from FIG. 26, for example, when the tilt angle reduction ratio is set to be 20%, the radiation amount in the core portion in which the tilt-angle is reduced decreases by about 10%, and the pressure loss therein decreases by about 25%. By this decreasing pressure loss decreasing by about 25%, the radiation amount in the core portion in which the tile-angle of the louver is the common angle for attaining the high heat transfer ratio is improved about 4%.

**[0096]** In the above described embodiments, the present invention is applied to the heat exchanger in which the condenser core portion 2 and the radiator core portion 3 are integrated. However, it is to be noted that the present invention can be applied to various heat exchangers in which two heat exchanging core portions, to carry out heat exchanges between two kinds of fluid and the air, are integrated.

**[0097]** Although the present invention has been fully described in connection with preferred embodiments thereof with reference to the accompanying drawings, it is to be noted that various changes and modifications will become apparent to those skilled in the art. Such changes and modifications are to be understood as being within the scope of the present invention as defined by the appended claims.

## Claims

### 1. A heat exchanger (1) comprising:

a first core portion (2) to carry out a heat exchange between a first fluid and an external fluid, said first core portion (2) including a plurality of first tubes (21) through which the first fluid flows and a first cooling fin (22) having plural louvers (220) disposed between said each pair of adjacent first tubes (21); and  
a second core portion (3) disposed to carry out a heat exchange between a second fluid and the external fluid, said second core portion (3) including a plurality of second tubes (31) through which the second fluid flows and a second cooling fin (32) having plural louvers (320) disposed between said each pair of adjacent second tubes (31); wherein  
said first core portion (2) and said second core portion (3) are disposed in parallel with a predetermined clearance (46) therebetween,  
said first cooling fin (22) and said second cooling fin (23) are integrated by a connecting portion (45), and  
a length of said louver in one core portion, out of said first and second core portions, a required radiation amount of which is smaller than that of the other core portion is shorter than a length of said louver in the other core portion.

2. A heat exchanger (1) according to claim 1, wherein the length of said louver in one core portion having a smaller required radiation amount than that in the other core portion is less than 50% of the length of said louver in the other core portion.

3. A heat exchanger (1) according to claim 1, wherein a tilt angle of said louver in said one core portion, out of said first and second core portions, the required radiation amount of which is smaller than that of the other core portion is smaller than a tilt angle of said louver in the other core portion.

4. A heat exchanger (1) according to claim 1, wherein said first core portion (2) is a condenser core portion (2) for condensing a refrigerant of a condenser for forming a refrigeration cycle,  
said second core portion (3) is a radiator core portion (3) for cooling an engine coolant of an automotive engine,  
said external fluid is cooling air for condensing the refrigerant and cooling the engine coolant, and  
said condenser core portion (2) is disposed at an air upstream side of said radiator core portion (3).

FIG. 1

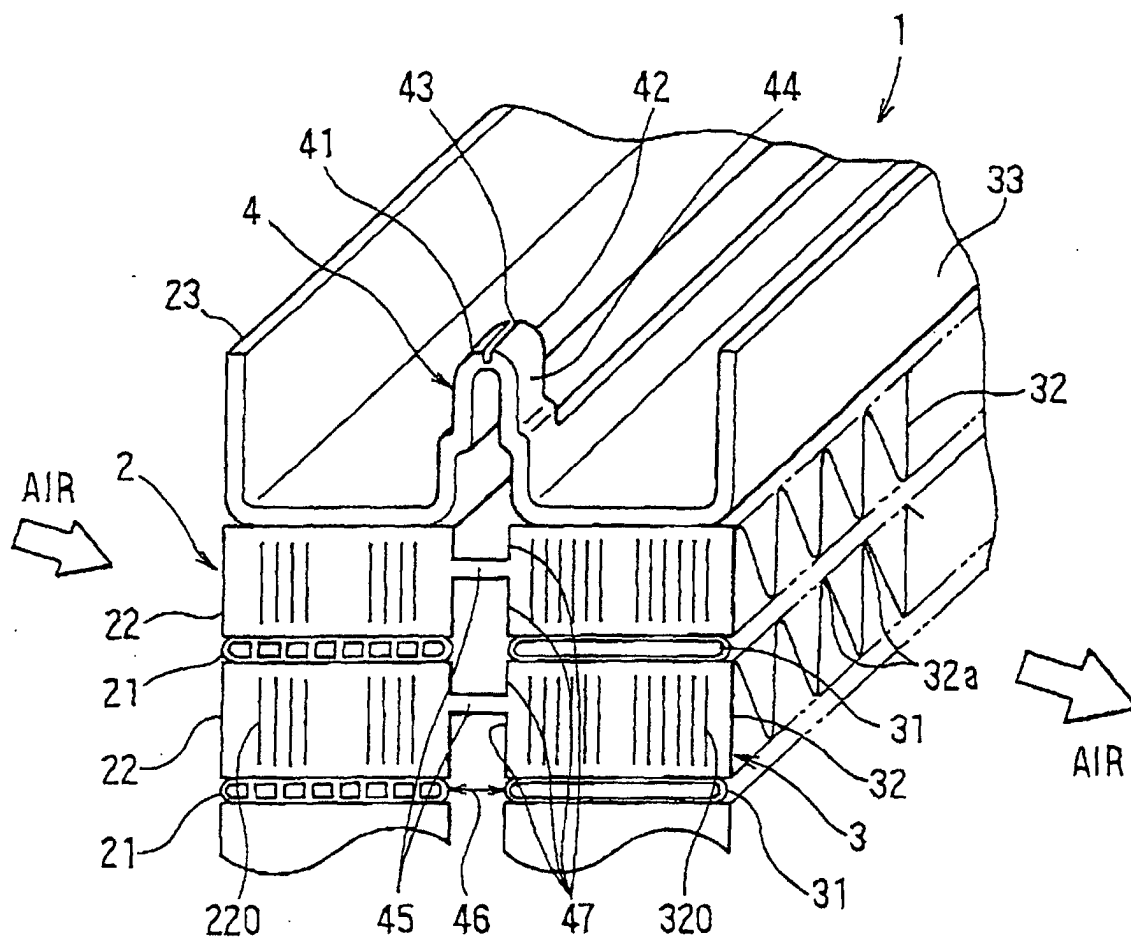


FIG. 2

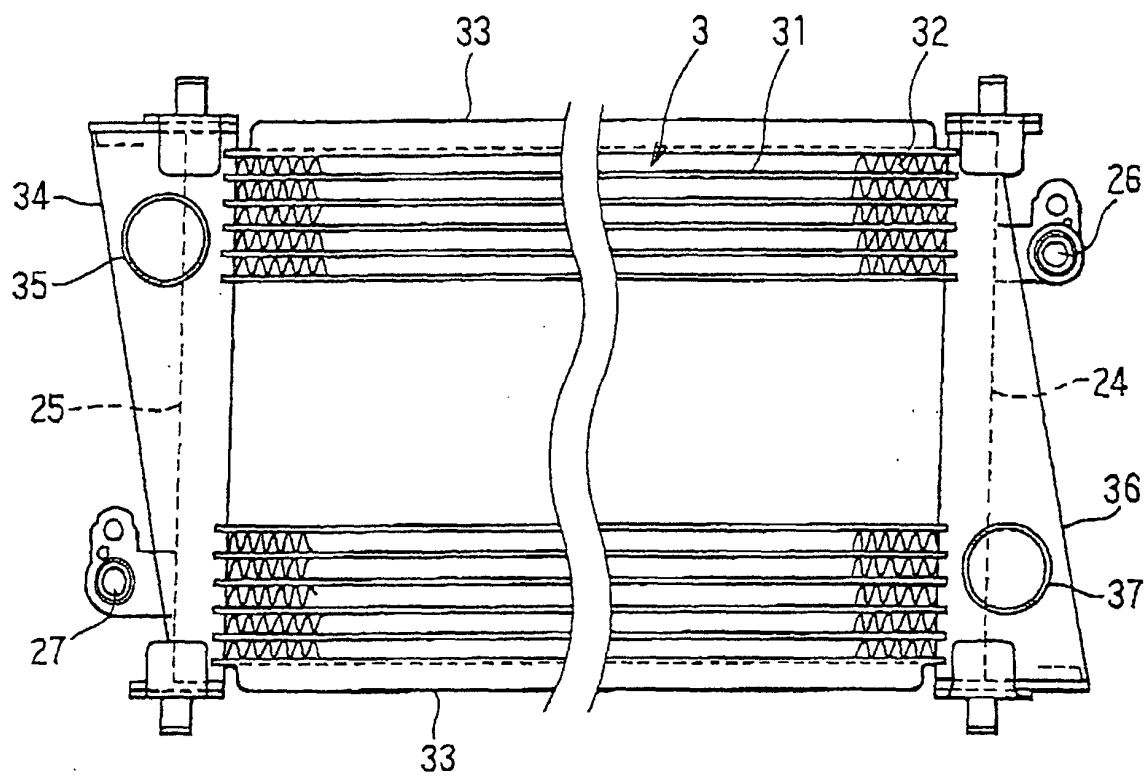


FIG. 3

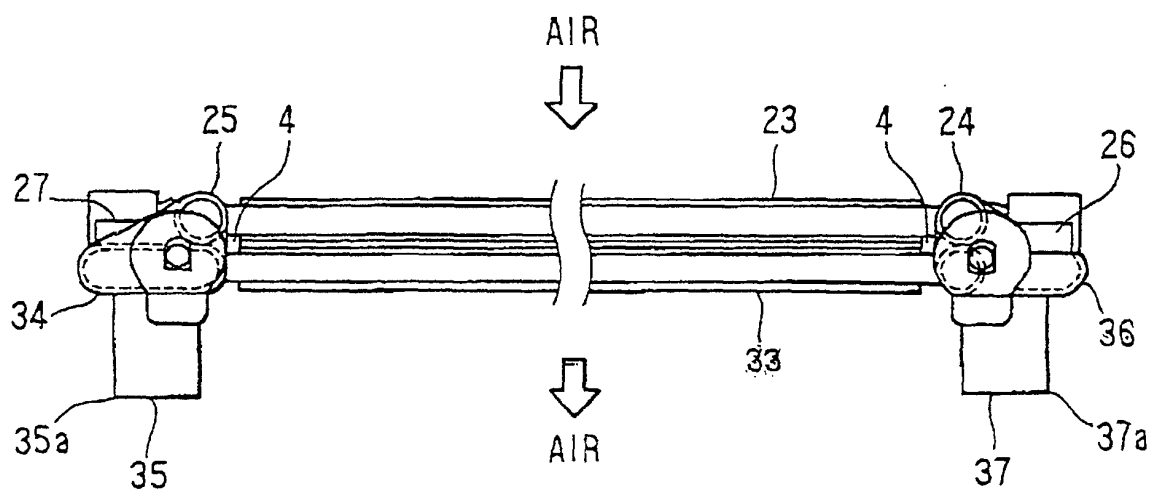


FIG. 4

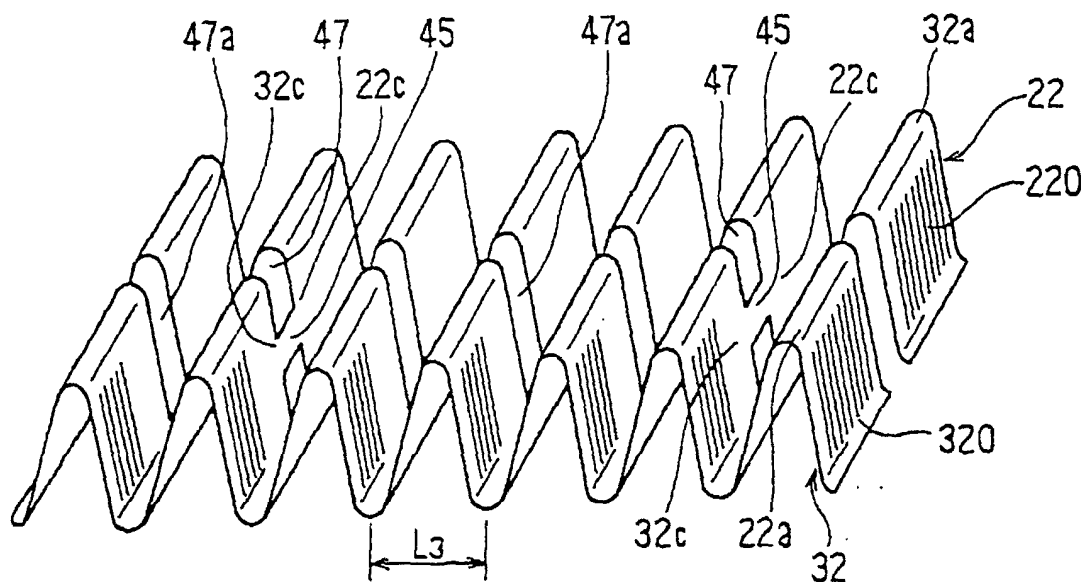


FIG. 5A

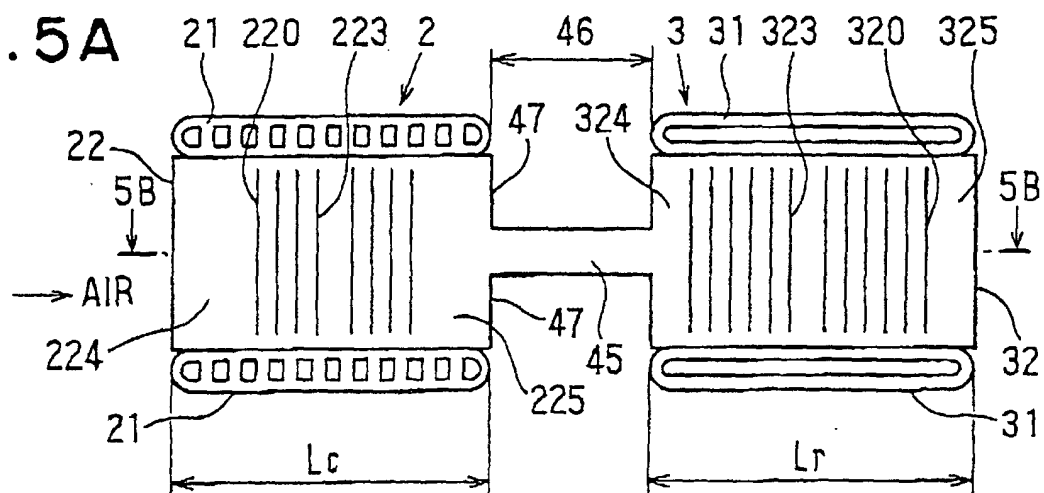


FIG. 5B

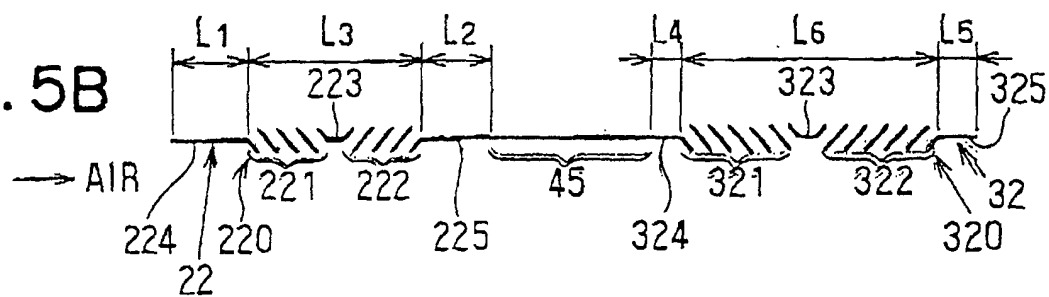


FIG. 6A

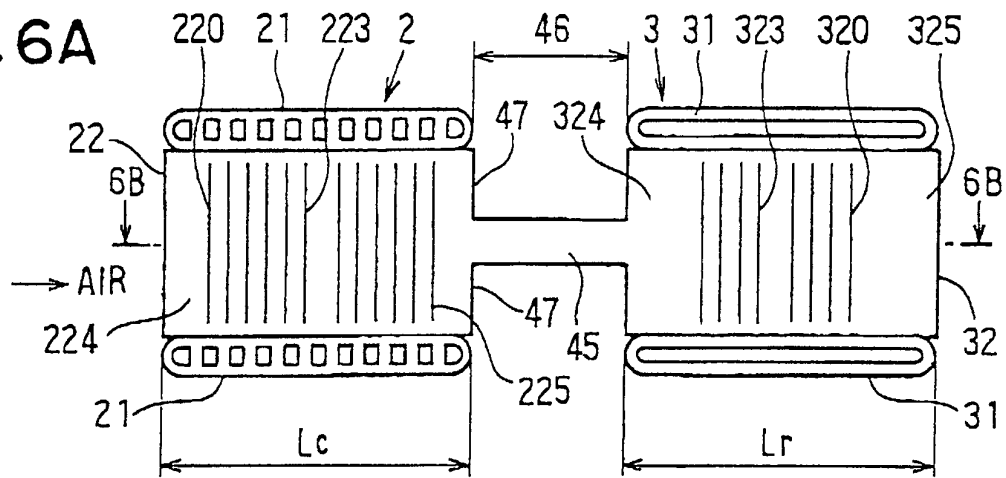


FIG. 6B

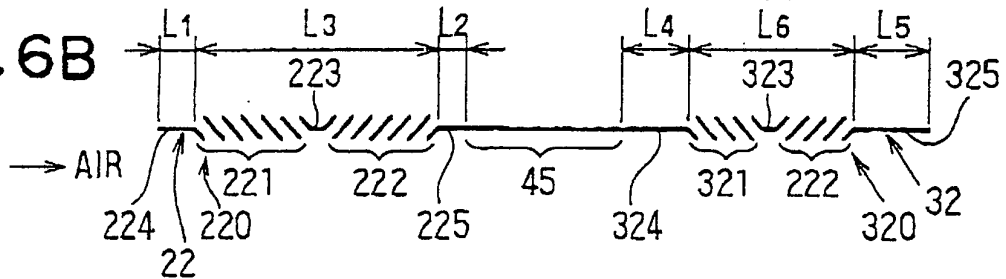


FIG. 7

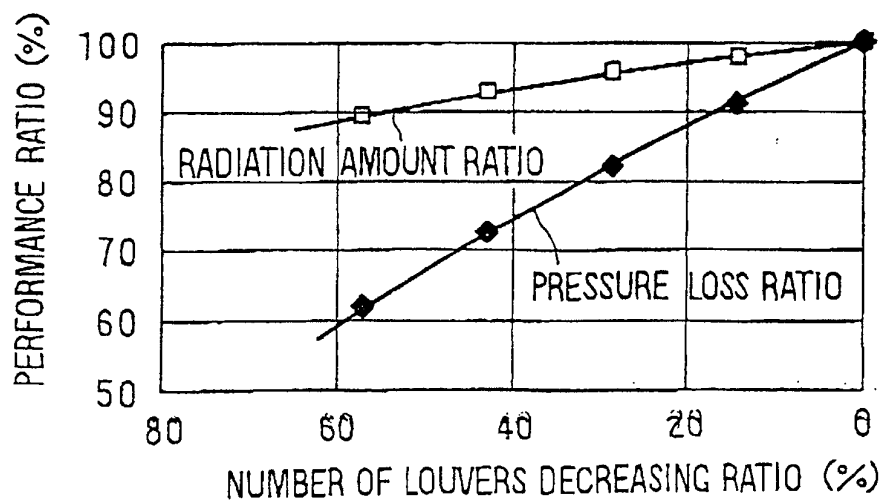


FIG. 8A

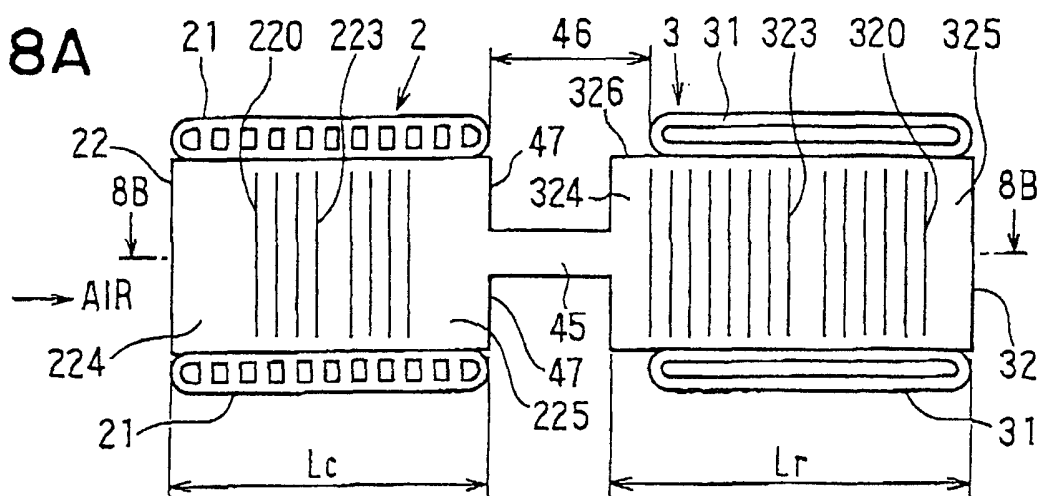


FIG. 8B

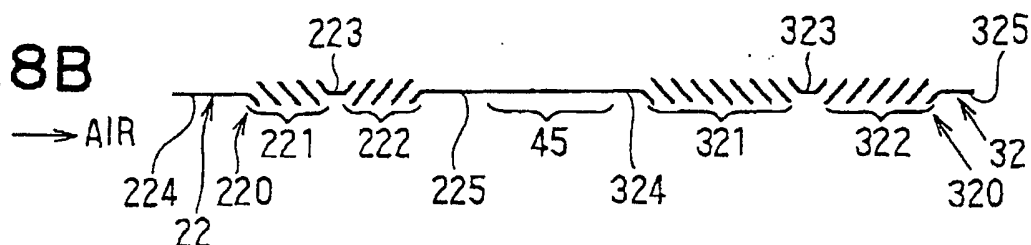


FIG. 9A

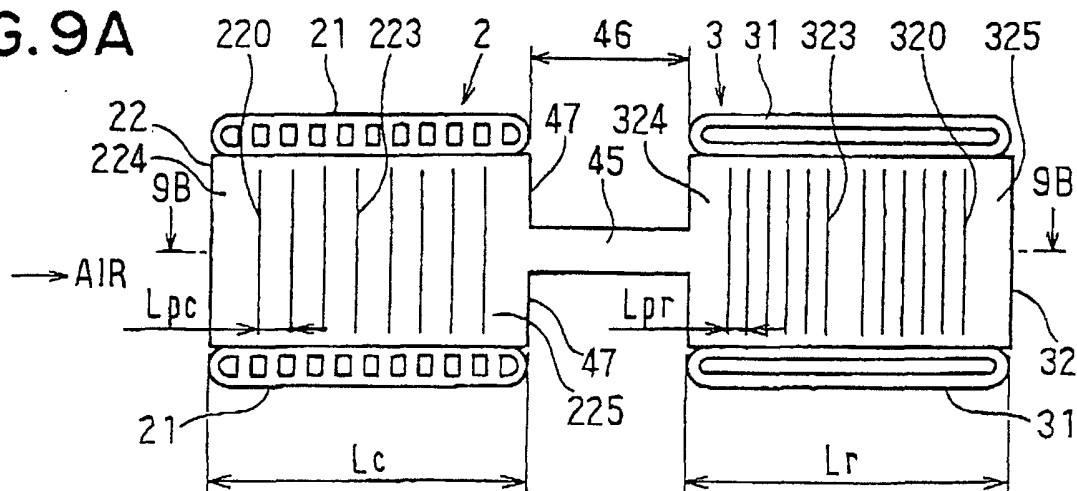


FIG. 9B

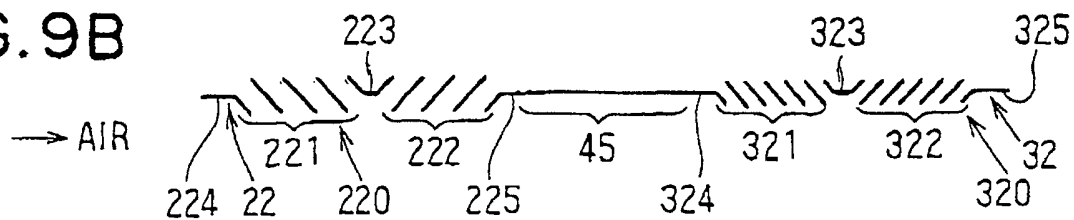


FIG.10A

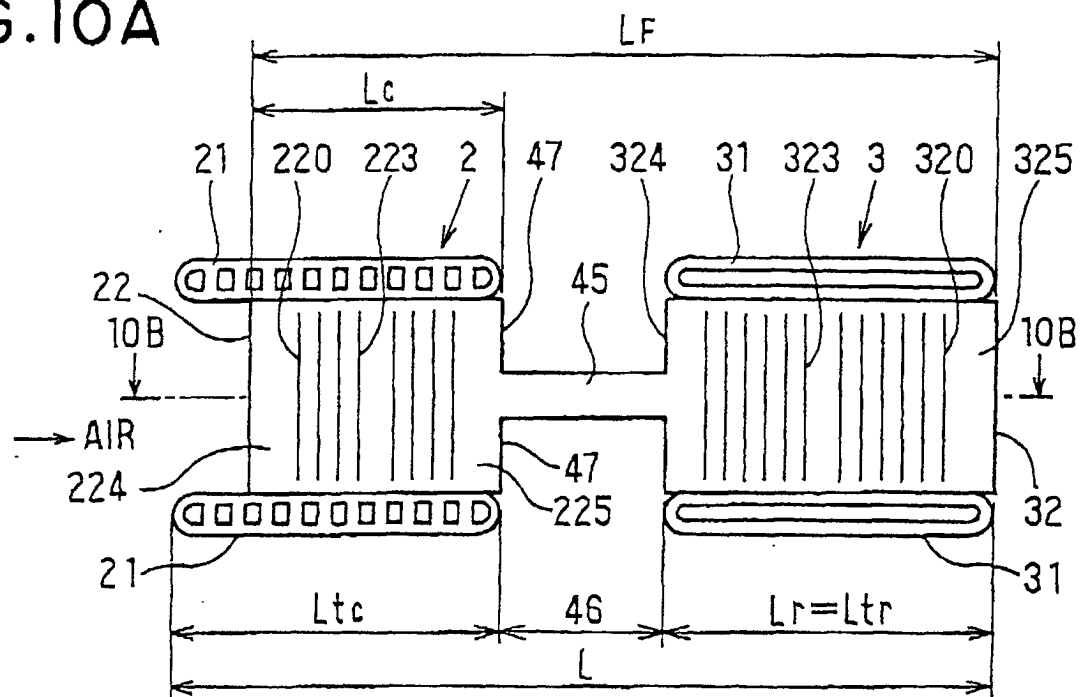


FIG.10B

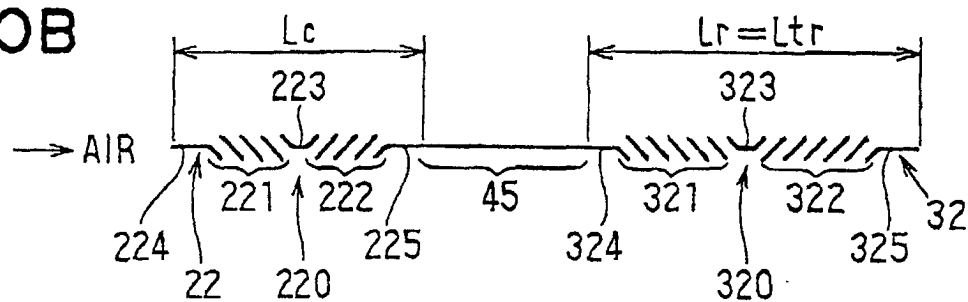




FIG. 11A

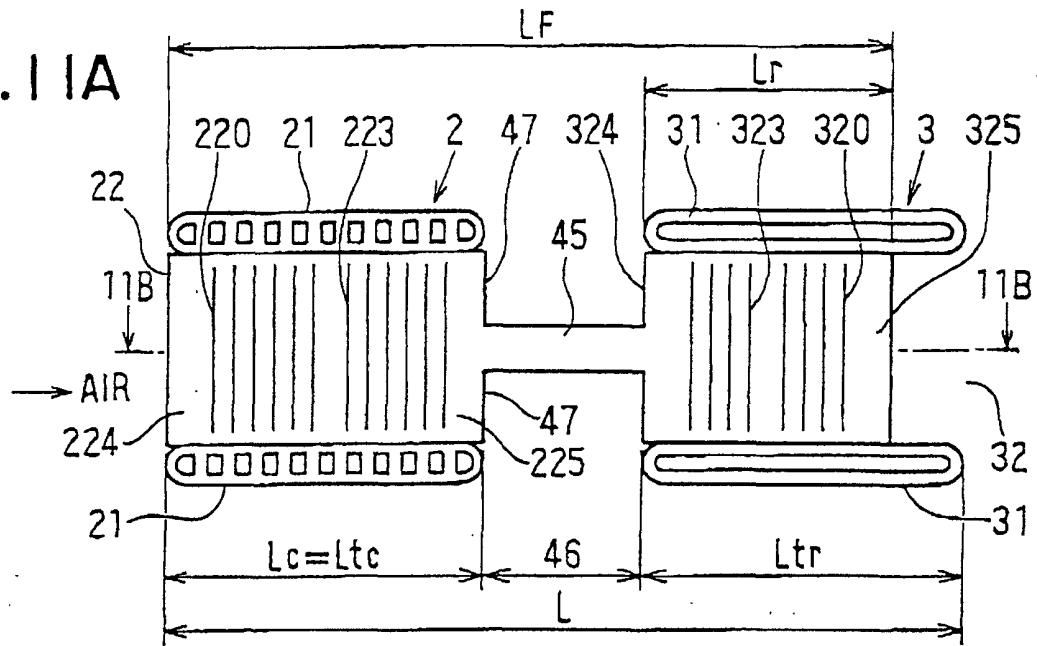


FIG. 11B

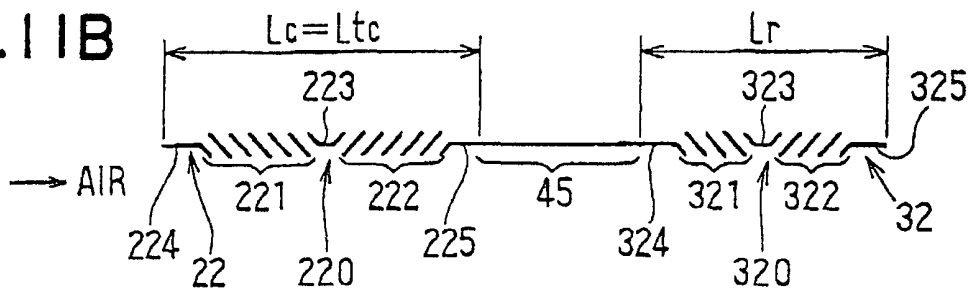


FIG. 12

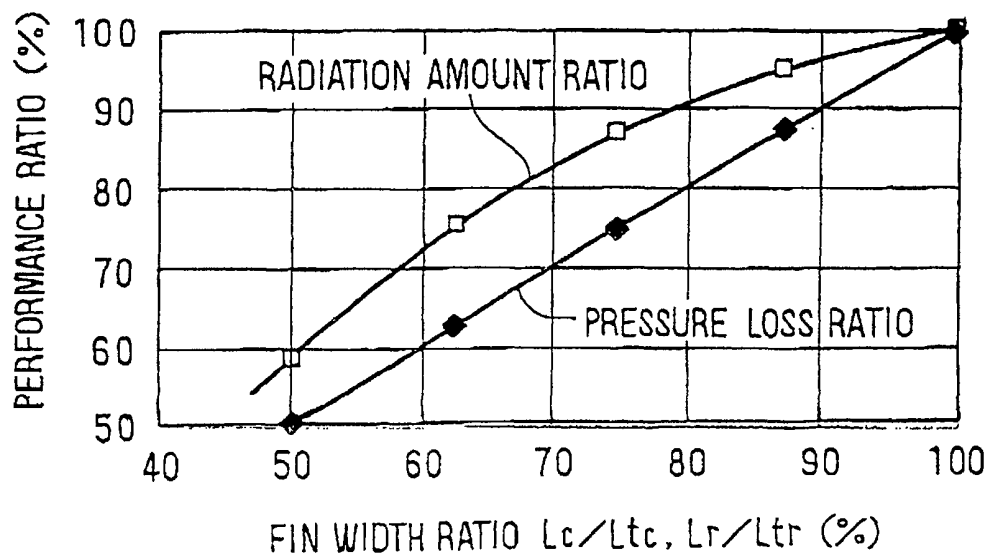


FIG. 13A

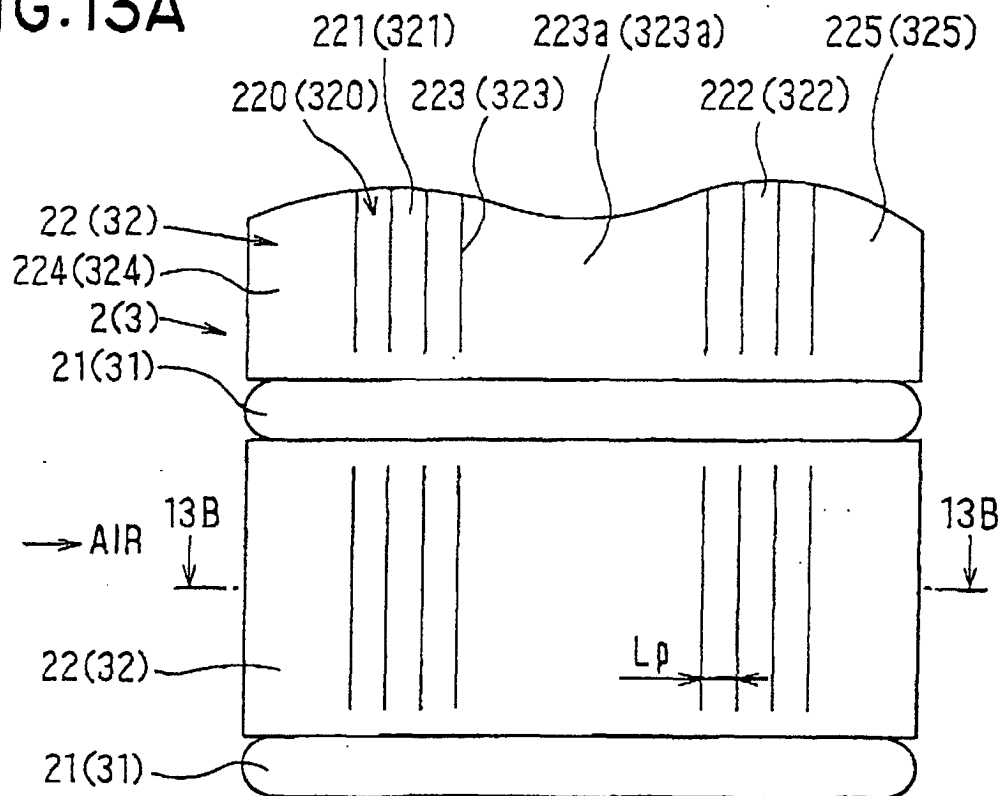


FIG. 13B

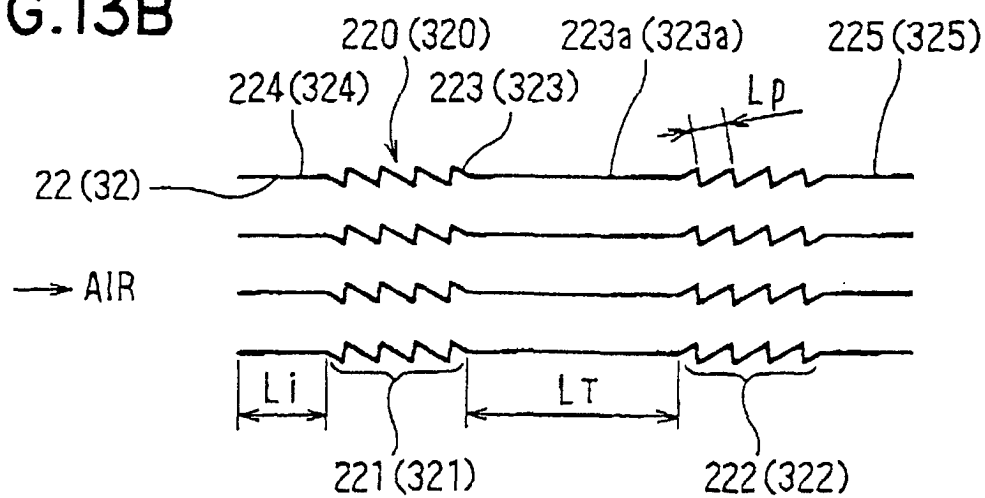


FIG. 14A

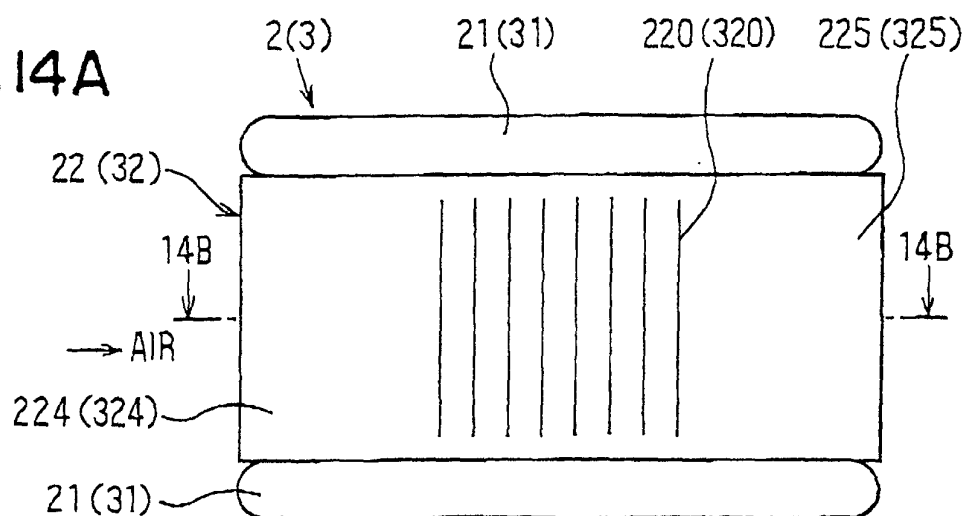


FIG. 14B

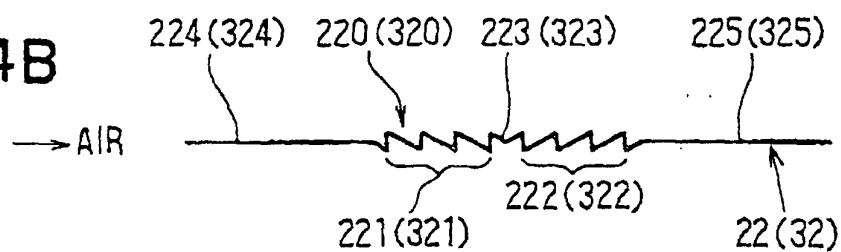


FIG. 15A

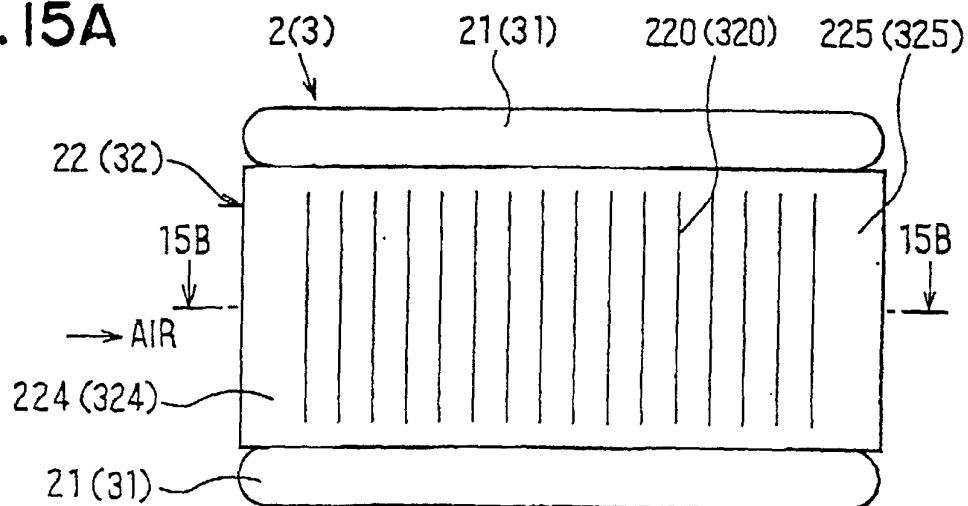


FIG. 15B

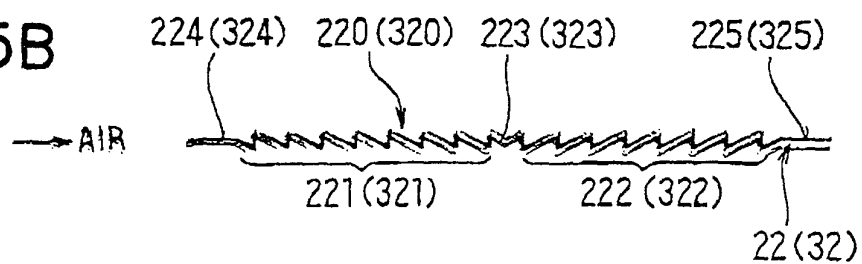


FIG. 16

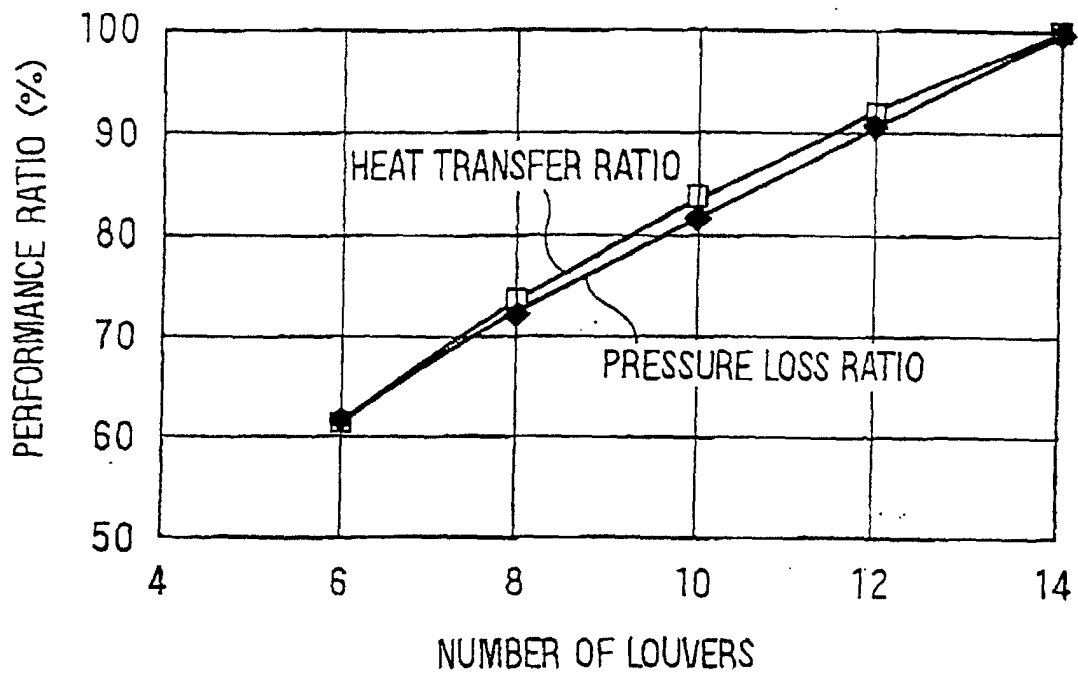


FIG. 17

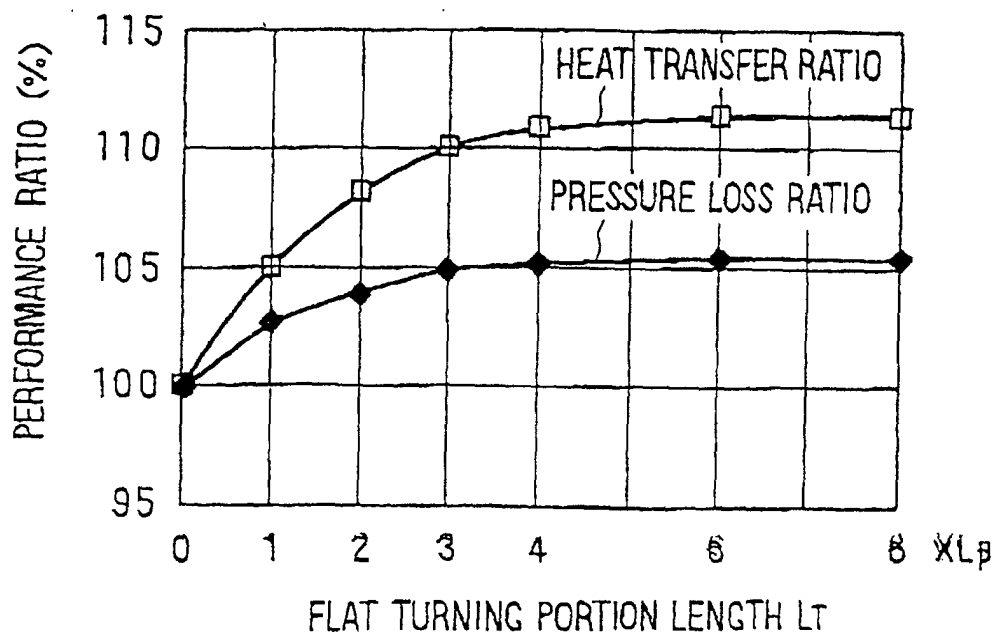


FIG. 18A

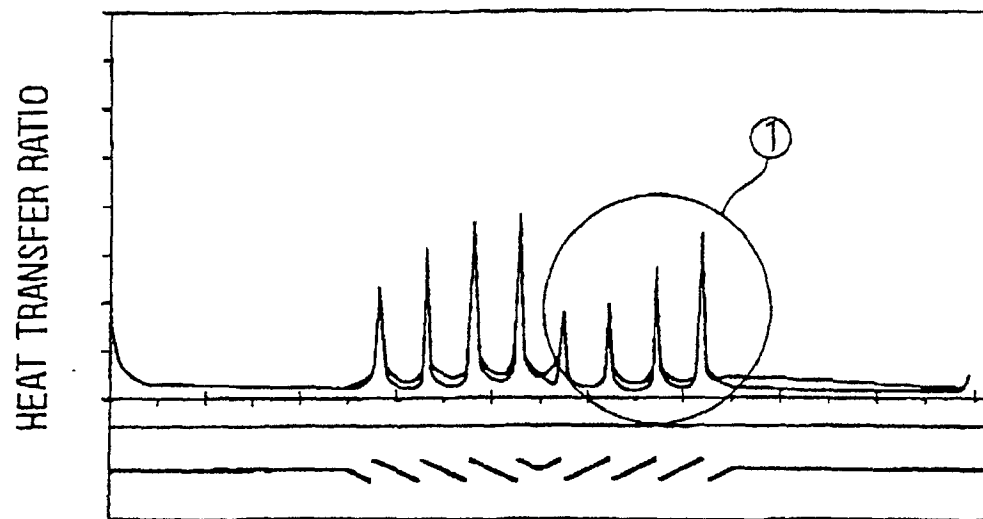


FIG. 18B

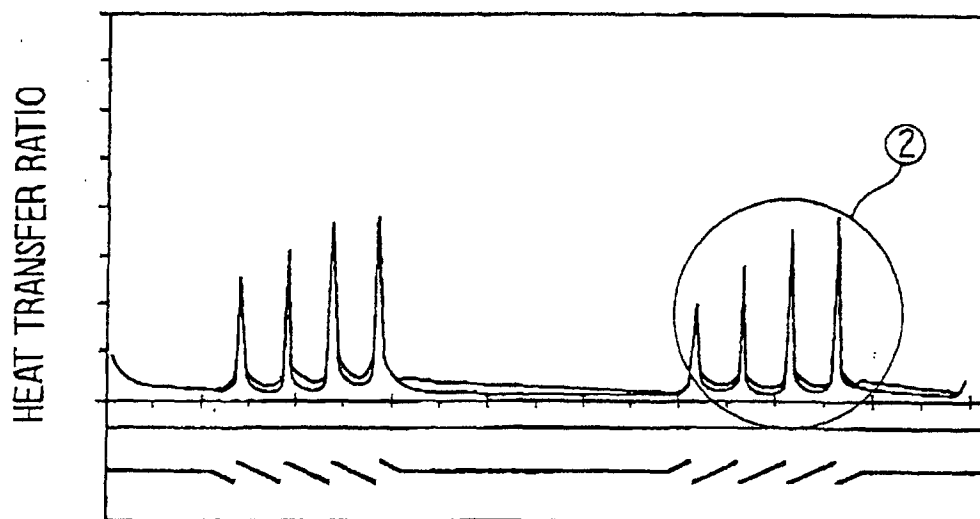


FIG. 19A

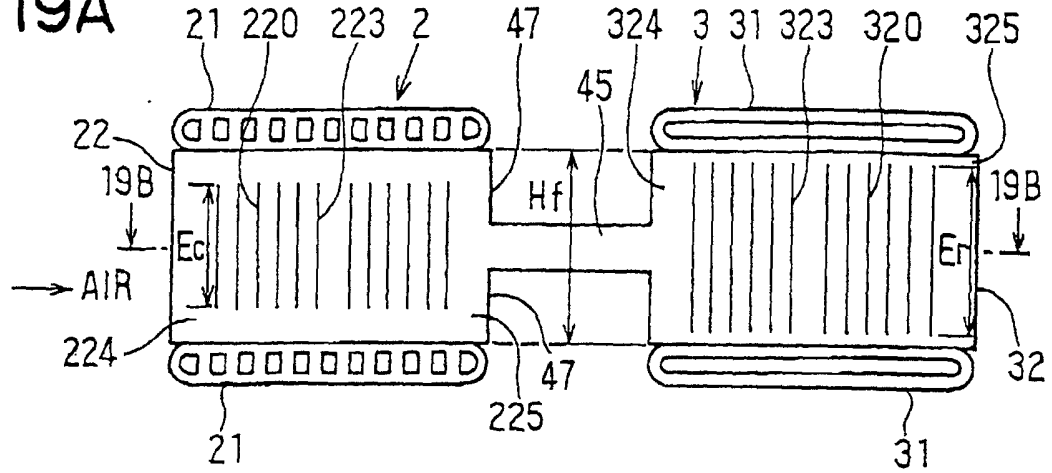


FIG. 19B

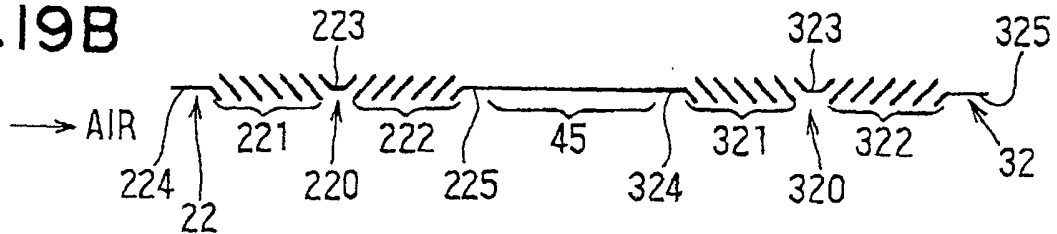


FIG. 20A

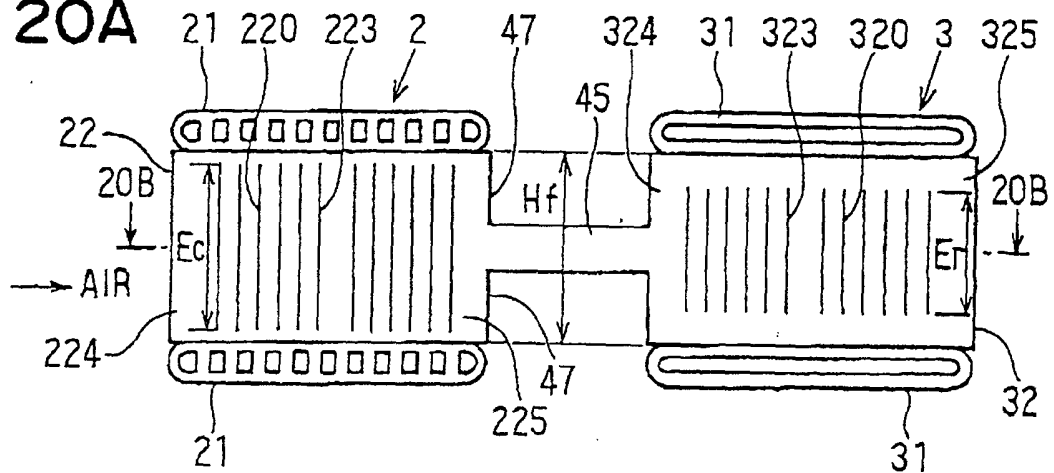
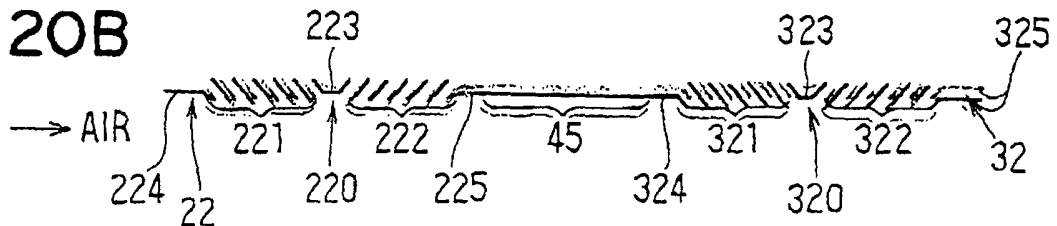
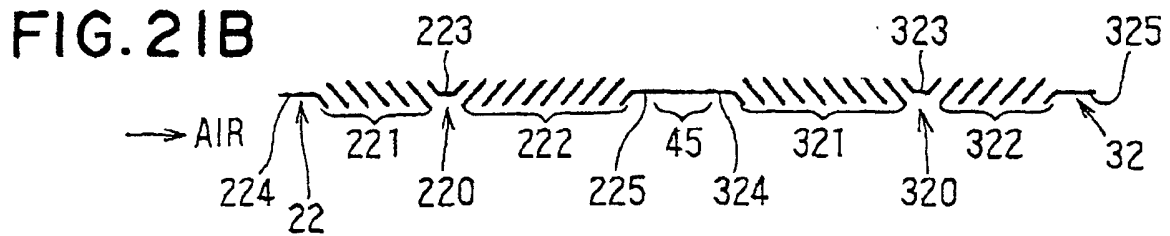
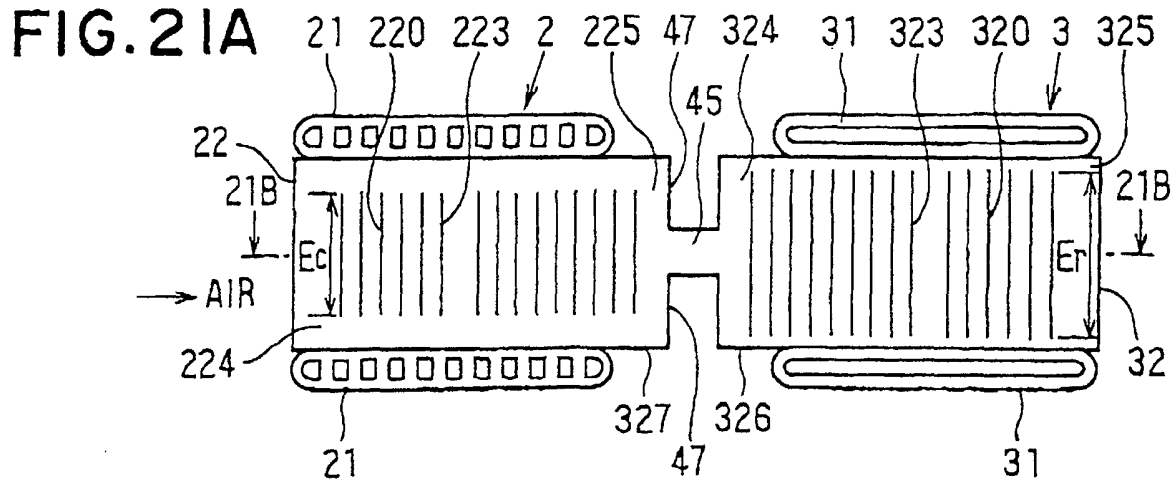


FIG. 20B





**FIG. 22**

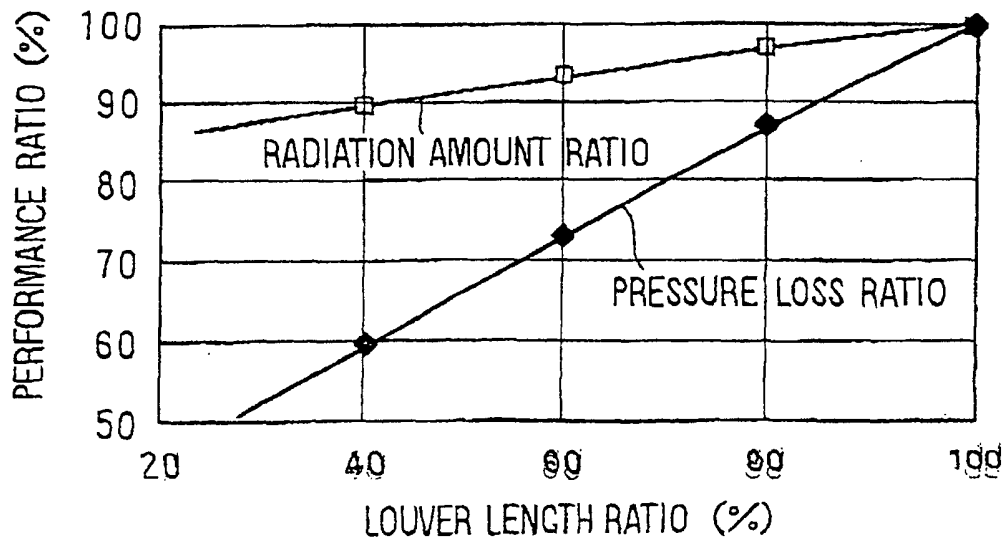


FIG. 23A

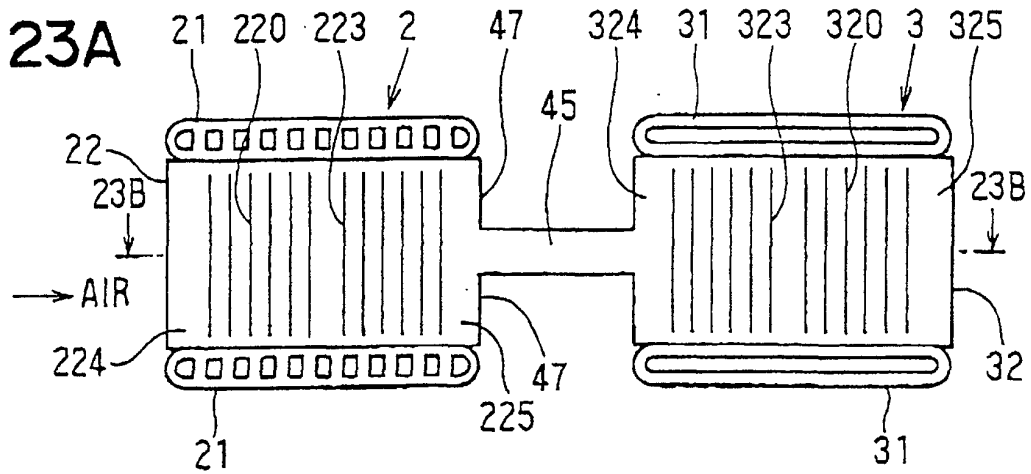


FIG. 23B

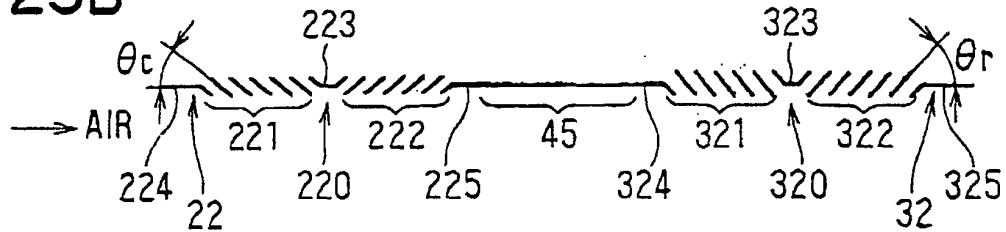


FIG. 24A

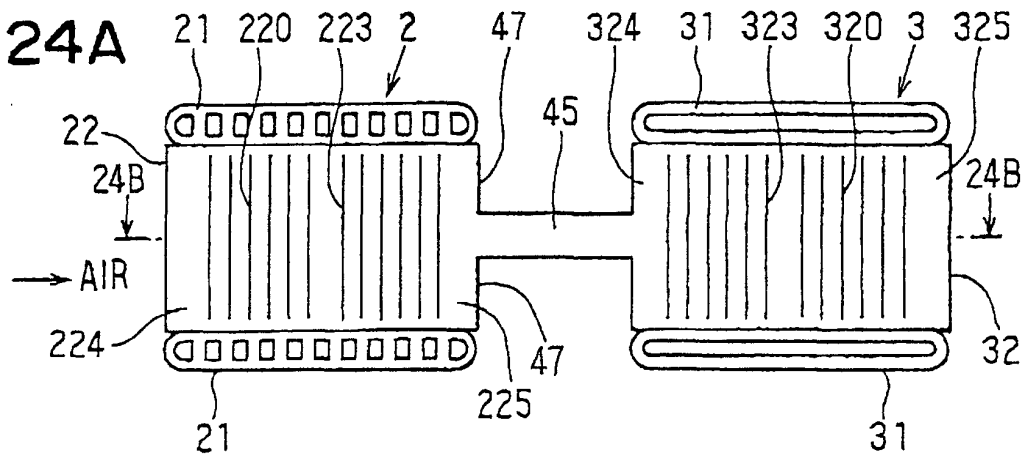


FIG. 24B

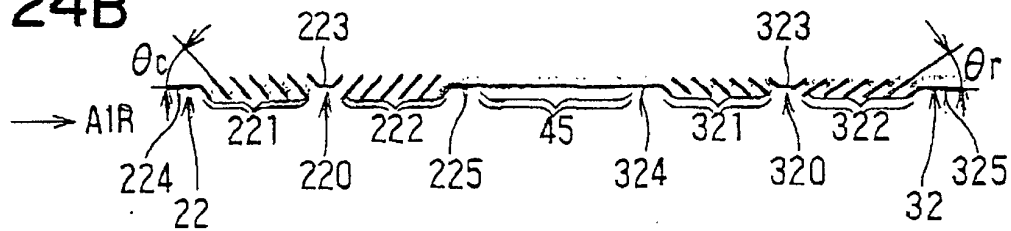




FIG. 25A

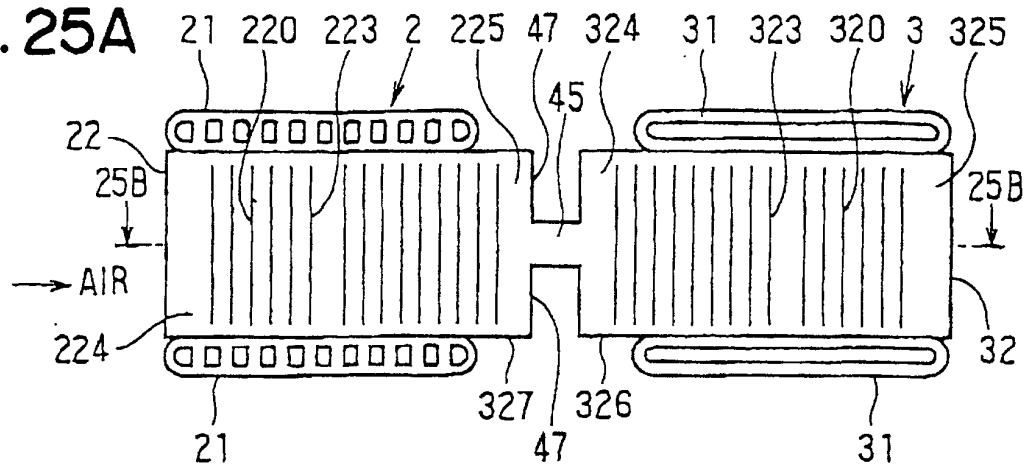


FIG. 25B

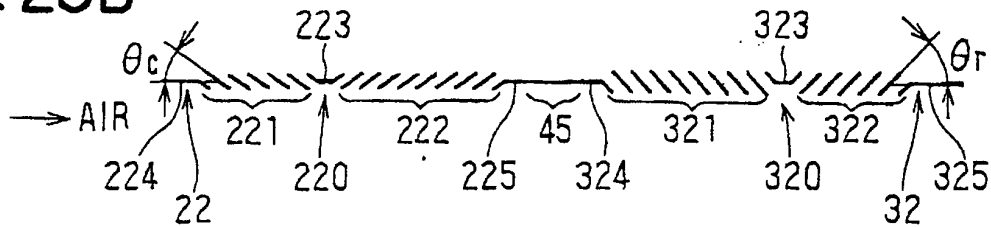
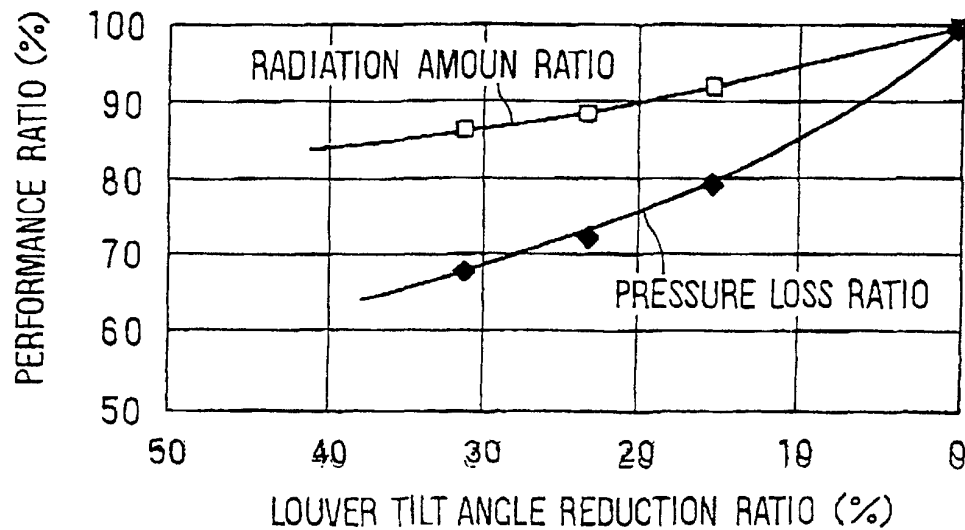


FIG. 26





European Patent  
Office

# EUROPEAN SEARCH REPORT

Application Number  
EP 02 00 0277

DOCUMENTS CONSIDERED TO BE RELEVANT			
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	CLASSIFICATION OF THE APPLICATION (Int.Cl.7)
E	EP 0 881 450 A (ZEXEL CORP) 2 December 1998 (1998-12-02) * column 6, line 47 - column 7, line 5; figures 2,3 *	1,4	F28D1/04 F28F1/12
A	US 5 529 116 A (WATANABE HIROHIKO ET AL) 25 June 1996 (1996-06-25) * column 12, line 5 - line 23; figures 27,43 *	1-4	
A	PATENT ABSTRACTS OF JAPAN vol. 018, no. 594 (M-1703), 14 November 1994 (1994-11-14) -& JP 06 221787 A (NIPPONDENSO CO LTD), 12 August 1994 (1994-08-12) * abstract; figures *	3	
A	EP 0 367 078 A (SANDEN CORP) 9 May 1990 (1990-05-09) * column 2, line 19 - column 3, line 1; figures 3,4 *	1-4	
			TECHNICAL FIELDS SEARCHED (Int.Cl.7)
			F28F F28D
The present search report has been drawn up for all claims			
Place of search <b>THE HAGUE</b>		Date of completion of the search <b>12 February 2002</b>	Examiner <b>Mootz, F</b>
CATEGORY OF CITED DOCUMENTS X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document			

EPO FORM 1503 03/82 (P04C01)

**ANNEX TO THE EUROPEAN SEARCH REPORT  
ON EUROPEAN PATENT APPLICATION NO.**

EP 02 00 0277

This annex lists the patent family members relating to the patent documents cited in the above-mentioned European search report. The members are as contained in the European Patent Office EDP file on  
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12-02-2002

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