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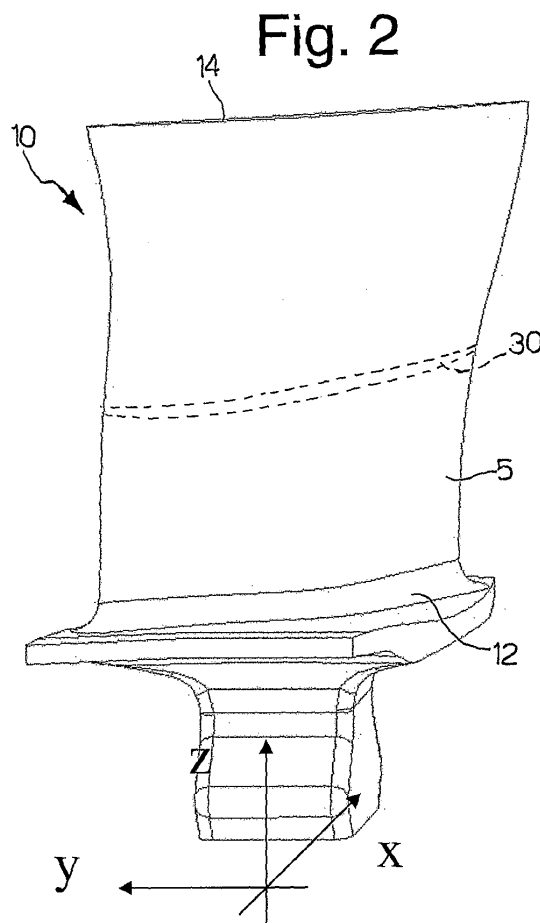
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(54) **Rotor blade for a ninth phase of a compressor**

(57) The invention relates to a blade (10) of a rotor of a ninth phase of a compressor, which can be defined by coordinates of a discrete combination of points, in a Cartesian reference system (X,Y,Z), wherein the axis (Z) is a radial axis intersecting the central axis of the compressor, said blade (10) having a profile which can be identified by means of a series of closed intersection curves between the profile itself and planes (X,Y) lying at distances (Z) from the central axis, said blade (10) also comprising a thickening (30), substantially parallel to a base portion (12) of the blade (10) itself, fixable to said rotor, said thickening (30) being substantially situated half-way up the blade (10) and being suitable for shifting the natural resonance frequencies of the blade (10) itself outside a functioning frequency range of said rotor.



**Description**

**[0001]** The present invention relates generally to a blade of a rotor of a ninth phase of a compressor.

**[0002]** More specifically, the invention relates to a blade of a rotor having a high aerodynamic efficiency of a ninth phase of a compressor.

**[0003]** Compressors normally pressurize in their interior air removed from the outside.

**[0004]** The fluid penetrates the compressor through a series of inlet ducts.

**[0005]** In these channels, the gas has low pressure and low temperature characteristics, whereas as it passes through the compressor, the gas is compressed and its temperature increases.

**[0006]** In order to increase the efficiency, the compressor is normally divided into various phases, each of which has a rotor and a stator respectively equipped with a series of blades.

**[0007]** In recent years, technologically advanced compressors have been further improved, obtaining an increased improvement in efficiency, operating in particular on the aerodynamic conditions.

**[0008]** The geometric configuration of the blades in fact significantly influences the aerodynamic efficiency.

**[0009]** This depends on the fact that the geometric characteristics of the blade cause a distribution of the relative velocities in the fluid, consequently influencing the distribution of the limit layers along the walls and, ultimately, losses due to friction.

**[0010]** In particular, in the case of rotor blades of a ninth phase of a compressor an extremely high efficiency is required, at the same time maintaining an appropriate aerodynamic and mechanical load.

**[0011]** An objective of the present invention is to provide a blade of a rotor of a ninth phase of a compressor which avoids or in any case reduces resonance problems due to flexural vibrations which reduce the life of the component, and at the same time allow a high aerodynamic efficiency.

**[0012]** A further objective of the present invention is to provide a rotor of a ninth phase of a compressor which allows a high aerodynamic efficiency and at the same time allows a high reliability of the compressor to be obtained with a consequent increase in the power of the turbine itself with the same compressor dimensions.

**[0013]** These objectives according to the present invention are addressed by providing a rotor blade of a ninth phase of a compressor as specified in claim 1.

**[0014]** Further characteristics of the invention are indicated in the subsequent claims.

**[0015]** The characteristics and advantages of a rotor blade of a ninth phase of a compressor according to various embodiments of the present invention will appear more evident from the following illustrative and non-limiting description, referring to the enclosed schematic drawings in which:

Figure 1 is a raised view of a rotor blade of a compressor produced with an aerodynamic profile according to an embodiment of the present invention;

Figure 2 is a raised view of the opposite side of the blade of figure 1; and

Figure 3 is a diagram of the maximum thickness trend of a blade according to an embodiment of the present invention, with respect to its height.

**[0016]** With reference to the figures, a blade 10 is provided of a rotor of a ninth phase of a compressor.

**[0017]** Said blade 10 is defined by means of coordinates of a discreet combination of points, in a Cartesian reference system (X,Y,Z), wherein the axis (Z) is a radial axis intersecting the central axis of the compressor, not shown.

**[0018]** The profile of the blade 10 is identified by means of a series of closed intersection curves between the profile itself and planes (X,Y) lying at distances (Z) from the central axis.

**[0019]** The profile of said blade 10 comprises a first substantially concave surface 3, which is pressurized, and a second substantially convex surface 5 which is in depression and opposite the first.

**[0020]** The two surfaces 3, 5 are continuous and joined to each other, and together form the profile of said blade 10.

**[0021]** At a base portion 12, commonly called "foot" of the blade 10, according to the known art there is a connecting joint with the aerodynamic profile of the blade 10 itself, said base portion 12 being suitable for being fixed to said rotor of said compressor.

**[0022]** Said blade 10 comprises a thickening 30, i.e. a prolonged portion having a greater thickness with respect to the adjacent portions, which is substantially parallel to said base portion 12 so as to shift the resonance frequencies of said blade 10 outside a functioning frequency range of the rotor itself, thus reducing or in any case avoiding problems of instability and vibrations of the blade 10 and rotor.

**[0023]** This advantageously leads to an increase in both the useful life and reliability of the rotor and compressor itself.

**[0024]** Said thickening 30 relates to at least one section or closed curve, and is also situated half-way up the blade 10.

**[0025]** In other words, said thickening 30 confers a dynamic behaviour to said blade which is such as to have flexural

frequencies which fall outside a functioning velocity range of the rotor of said compressor and consequently such that there is no intensification of the maximum flexural deformation of the blade during the functioning of the compressor.

**[0026]** This consequently leads to a higher performance of the compressor, of the rotor and a longer useful life of its components, as problems of resonance such as those described above are avoided.

**[0027]** The clearances and tolerances of the blade and stator can therefore be dimensioned so as to further increase the performances of the compressor itself.

**[0028]** This is possible as the blade 10 is prevented, upon deforming, from causing a contact and relative friction against the relative stator.

**[0029]** In particular, each closed curve has a maximum thickness determined by the maximum distance between said first surface 3 and said second surface 5.

**[0030]** Said maximum surface of each closed curve, along the height of the blade 10, moving towards a free end 14 of the blade 10, has first a decreasing and then an increasing trend, followed again by a decreasing and finally increasing trend, with two different slopes, said blade 10 comprising a further thickening substantially parallel to said base portion 12 and situated in particular close to said free end 14.

**[0031]** For example, the variation in the trend of the maximum thickness is shown in figure 3, in which it is compared with the maximum thickness trend of a blade according to the known art. In particular, in figure 3, the abscissa indicates the height of the blade 10, whereas the ordinate represents the maximum thickness of the blade 10, adimensionalized by putting the thickness in correspondence with the foot of the blade equal to 1. In the diagram shown in figure 3, the lower line represents the maximum thickness trend of a blade according to the known art, whereas the upper line shows the trend of the maximum thickness of the blade according to an embodiment of the present invention.

**[0032]** Along the height of the blade 10 in the direction of a free end 14 of the blade 10, said maximum thickness preferably has a trend which can be described by four different mathematical functions, identifying four different regions of the blade.

**[0033]** In the first region, that closest to the blade 10, up to a height equal to 45% of the height of the blade, the maximum thickness trend can be described by a polynomial function of the fourth degree (first decreasing and subsequently increasing) and in particular said polynomial function is:

$$T_{\max} = -34.522 \cdot h^4 + 36.4 \cdot h^3 - 8.4113 \cdot h^2 - 0.7259 \cdot h + 0.9961$$

wherein h represents the percentage of the height of the blade 10, and wherein T<sub>max</sub> is the maximum adimensionalized thickness relating to that closed curve corresponding to that percentage of the height of the blade 10.

**[0034]** In the subsequent region, ranging from 45% to 58% of the height of the blade 10, the thickness varies according to the linear function (decreasing):

$$T_{\max} = -1.3509 \cdot h + 1.4459$$

**[0035]** Therefore, between 58% and 86% of the height of the blade 10, the thickness trend is represented by the linear function (increasing):

$$T_{\max} = 0.2074 \cdot h + 0.5443$$

**[0036]** Finally, between 86% and the free end 14 of the blade, the maximum thickness varies according to the linear function (increasing):

$$T_{\max} = 0.9058 \cdot h - 0.0518$$

**[0037]** The profile of each blade 10 was also suitably shaped to be able to maintain the same efficiency at high levels.

**[0038]** The aerodynamic profile of each blade 10 is preferably defined by means of a series of closed curves whose coordinates are defined with respect to a Cartesian reference system X, Y, Z, wherein the axis Z is a radial axis intersecting the central axis of the turbine, and said closed curves lying at distances Z from the central axis are defined according

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to Table I, whose values, expressed in millimeters, refer to an aerodynamic profile at room temperature, in particular 25°C.

Table I

X	Y	Z
-8.852	9.902	255.999
-8.847	9.904	255.999
-8.835	9.905	255.999
-8.812	9.901	255.999
-8.771	9.880	255.999
-8.717	9.833	255.999

-8.632	9.734	255.999
-8.533	9.594	255.999
-8.410	9.400	255.999
-8.263	9.152	255.999
-8.080	8.826	255.999
-7.871	8.449	255.999
-7.643	8.050	255.999

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-7.382	7.603	255.999
-7.086	7.110	255.999
-6.751	6.573	255.999
-6.390	6.019	255.999
-6.005	5.448	255.999
-5.593	4.860	255.999
-5.155	4.256	255.999
-4.688	3.638	255.999
-4.190	3.006	255.999
-3.660	2.364	255.999
-3.096	1.711	255.999
-2.520	1.069	255.999
-1.933	0,30278	255.999
-1.338	-0.189	255.999
-0.735	-0.806	255.999
-0.126	-1.417	255.999
0,33958	-2.023	255.999
1.108	-2.624	255.999
1.732	-3.219	255.999
2.362	-3.809	255.999
2.995	-4.395	255.999
3.631	-4.978	255.999
4.249	-5.538	255.999

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4.849	-6.075	255.999
5.432	-6.588	255.999
5.998	-7.077	255.999
6.545	-7.543	255.999
7.075	-7.986	255.999
7.586	-8.405	255.999
8.056	-8.784	255.999
8.485	-9.124	255.999
8.871	-9.424	255.999
9.214	-9.685	255.999
9.515	-9.908	255.999
9.771	-10.093	255.999
9.992	-10.249	255.999
10.179	-10.379	255.999
10.332	-10.485	255.999
10.426	-10.600	255.999
10.461	-10.708	255.999
10.465	-10.793	255.999
10.452	-10.864	255.999
10.432	-10.917	255.999
10.411	-10.955	255.999
10.385	-10.991	255.999
10.343	-11.034	255.999
10.282	-11.077	255.999

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10.199	-11.110	255.999
10.081	-11.119	255.999
9.933	-11.073	255.999
9.765	-10.976	255.999
9.561	-10.857	255.999
9.318	-10.717	255.999
9.034	-10.552	255.999
8.697	-10.359	255.999
8.308	-10.137	255.999
7.867	-9.884	255.999
7.376	-9.600	255.999
6.835	-9.282	255.999
6.245	-8.931	255.999
5.633	-8.559	255.999
4.999	-8.165	255.999
4.344	-7.748	255.999
3.670	-7.305	255.999
2.977	-6.837	255.999
2.267	-6.341	255.999
1.543	-5.816	255.999
0,57639	-5.280	255.999
0,08958	-4.730	255.999
-0.558	-4.165	255.999

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-1.230	-3.583	255.999
-1.886	-2.983	255.999
-2.526	-2.366	255.999
-3.146	-1.727	255.999
-3.744	-1.064	255.999
-4.317	-0.373	255.999
-4.864	0,24028	255.999
-5.383	1.085	255.999
-5.861	1.816	255.999
-6.301	2.535	255.999
-6.704	3.242	255.999
-7.070	3.934	255.999
-7.401	4.611	255.999
-7.697	5.271	255.999
-7.962	5.912	255.999
-8.194	6.533	255.999
-8.387	7.104	255.999
-8.547	7.623	255.999
-8.675	8.088	255.999
-8.783	8.528	255.999
-8.864	8.911	255.999
-8.914	9.209	255.999
-8.940	9.449	255.999

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5	-8.945	9.630	255.999
	-8.932	9.765	255.999
	-8.911	9.837	255.999
10	-8.887	9.878	255.999
	-8.869	9.894	255.999
15	-8.858	9.900	255.999
	-8.675	9.767	258.001
20	-8.669	9.769	258.001
	-8.658	9.770	258.001
25	-8.635	9.767	258.001
	-8.593	9.746	258.001
30	-8.538	9.700	258.001
	-8.452	9.604	258.001
35	-8.349	9.465	258.001
	-8.222	9.274	258.001
40	-8.069	9.031	258.001
	-7.877	8.711	258.001
45	-7.657	8.340	258.001
	-7.419	7.947	258.001
50	-7.147	7.508	258.001
	-6.840	7.022	258.001
55	-6.494	6.493	258.001
	-6.123	5.948	258.001

-5.726	5.385	258.001
-5.304	4.805	258.001
-4.855	4.211	258.001
-4.377	3.601	258.001
-3.871	2.978	258.001
-3.333	2.342	258.001
-2.766	1.694	258.001
-2.187	1.055	258.001
-1.600	0,29514	258.001
-1.004	-0.198	258.001
-0.402	-0.814	258.001
0,14236	-1.425	258.001
0,56736	-2.031	258.001
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2.053	-3.232	258.001
2.677	-3.827	258.001
3.304	-4.418	258.001
3.933	-5.006	258.001
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5.139	-6.113	258.001
5.715	-6.632	258.001
6.274	-7.127	258.001
6.815	-7.600	258.001

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7.337	-8.049	258.001
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9.109	-9.512	258.001
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10.749	-10.863	258.001
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10.482	-11.152	258.001
10.366	-11.127	258.001
10.233	-11.049	258.001

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10.067	-10.949	258.001
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9.626	-10.684	258.001
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5.364	-8.089	258.001
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4.052	-7.219	258.001
3.368	-6.746	258.001
2.667	-6.245	258.001
1.952	-5.716	258.001
1.247	-5.175	258.001
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-0.127	-4.057	258.001
-0.794	-3.477	258.001
-1.448	-2.882	258.001
-2.087	-2.272	258.001



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	-3.895	-0.323	258.001
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	-4.991	1.094	258.001
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20	-6.356	3.203	258.001
	-6.736	3.884	258.001
25	-7.080	4.549	258.001
	-7.389	5.198	258.001
30	-7.668	5.828	258.001
	-7.915	6.439	258.001
35	-8.123	7.002	258.001
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40	-8.438	7.971	258.001
	-8.559	8.404	258.001
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	-8.713	9.076	258.001
50	-8.746	9.313	258.001
	-8.757	9.493	258.001
55	-8.749	9.627	258.001
	-8.731	9.700	258.001

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-8.691	9.759	258.001
-8.680	9.765	258.001
-8.508	9.651	259.194
-8.503	9.652	259.194
-8.491	9.654	259.194
-8.468	9.651	259.194
-8.427	9.631	259.194
-8.371	9.586	259.194
-8.283	9.491	259.194
-8.177	9.355	259.194
-8.046	9.168	259.194
-7.887	8.929	259.194
-7.688	8.614	259.194
-7.459	8.249	259.194
-7.212	7.863	259.194
-6.931	7.429	259.194
-6.615	6.950	259.194
-6.261	6.428	259.194
-5.882	5.888	259.194
-5.480	5.331	259.194
-5.052	4.757	259.194
-4.598	4.166	259.194

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5	-4.117	3.561	259.194
	-3.607	2.941	259.194
	-3.069	2.307	259.194
10	-2.502	1.660	259.194
	-1.925	1.022	259.194
15	-1.339	0,27222	259.194
	-0.745	-0.231	259.194
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	0,32083	-1.458	259.194
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	1.687	-2.667	259.194
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	2.925	-3.862	259.194
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	4.175	-5.044	259.194
40	4.784	-5.611	259.194
	5.374	-6.155	259.194
45	5.946	-6.677	259.194
	6.501	-7.176	259.194
50	7.037	-7.652	259.194
	7.555	-8.105	259.194
55	8.055	-8.535	259.194
	8.515	-8.924	259.194

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9.311	-9.582	259.194
9.646	-9.851	259.194
9.940	-10.081	259.194
10.190	-10.273	259.194
10.406	-10.435	259.194
10.588	-10.570	259.194
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10.860	-10.766	259.194
10.943	-10.845	259.194
10.975	-10.924	259.194
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10.915	-11.125	259.194
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10.798	-11.187	259.194
10.709	-11.185	259.194
10.603	-11.133	259.194
10.472	-11.052	259.194
10.309	-10.951	259.194
10.110	-10.828	259.194
9.873	-10.683	259.194

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10	8.456	-9.824	259.194
	7.977	-9.531	259.194
15	7.449	-9.204	259.194
	6.873	-8.842	259.194
20	6.275	-8.460	259.194
	5.655	-8.057	259.194
25	5.015	-7.630	259.194
	4.355	-7.180	259.194
30	3.678	-6.705	259.194
	2.985	-6.205	259.194
35	2.278	-5.676	259.194
	1.581	-5.137	259.194
40	0,62222	-4.586	259.194
	0,15417	-4.023	259.194
45	-0.440	-3.446	259.194
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	-2.340	-1.626	259.194
55	-2.941	-0.984	259.194
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15	-8.198	9.503	260.388
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	-7.867	9.090	260.388
25	-7.705	8.854	260.388
	-7.499	8.543	260.388
30	-7.264	8.183	260.388
	-7.010	7.801	260.388
35	-6.723	7.373	260.388
	-6.401	6.899	260.388
40	-6.041	6.381	260.388
	-5.658	5.845	260.388
45	-5.252	5.291	260.388
	-4.820	4.720	260.388
50	-4.364	4.133	260.388
	-3.880	3.529	260.388
55	-3.370	2.911	260.388
	-2.831	2.278	260.388

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1.906	-2.701	260.388
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3.139	-3.898	260.388
3.760	-4.492	260.388
4.384	-5.083	260.388
4.990	-5.651	260.388
5.578	-6.197	260.388
6.147	-6.720	260.388
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7.232	-7.699	260.388
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9.118	-9.328	260.388
9.494	-9.638	260.388
9.827	-9.909	260.388

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7.540	-10.004	271.165
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-0.758	-2.415	288.000
-0.214	-3.194	288.000
0,23125	-3.970	288.000
0,61319	-4.743	288.000
1.435	-5.516	288.000
1.990	-6.286	288.000
2.547	-7.055	288.000
3.107	-7.822	288.000
3.671	-8.585	288.000
4.220	-9.321	288.000
4.754	-10.028	288.000
5.273	-10.707	288.000
5.776	-11.358	288.000
6.263	-11.982	288.000
6.735	-12.577	288.000
7.190	-13.145	288.000
7.611	-13.660	288.000

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7.994	-14.123	288.000
8.341	-14.535	288.000
8.649	-14.896	288.000
8.919	-15.206	288.000
9.151	-15.466	288.000
9.351	-15.686	288.000
9.521	-15.869	288.000
9.662	-16.020	288.000
9.775	-16.139	288.000
9.860	-16.232	288.000
9.896	-16.321	288.000
9.893	-16.401	288.000
9.869	-16.461	288.000
9.840	-16.500	288.000
9.803	-16.531	288.000
9.744	-16.558	288.000
9.663	-16.563	288.000
9.573	-16.528	288.000
9.477	-16.442	288.000
9.354	-16.328	288.000
9.199	-16.186	288.000
9.011	-16.014	288.000
8.788	-15.810	288.000

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8.525	-15.571	288.000
8.215	-15.289	288.000
7.857	-14.963	288.000
7.453	-14.593	288.000
7.003	-14.177	288.000
6.510	-13.713	288.000
5.973	-13.202	288.000
5.417	-12.662	288.000
4.845	-12.094	288.000
4.258	-11.498	288.000
3.657	-10.872	288.000
3.043	-10.215	288.000
2.418	-9.527	288.000
1.784	-8.807	288.000
1.161	-8.078	288.000
0,38194	-7.339	288.000
-0.048	-6.591	288.000
-0.632	-5.833	288.000
-1.203	-5.065	288.000
-1.762	-4.287	288.000
-2.309	-3.501	288.000
-2.843	-2.705	288.000
-3.364	-1.899	288.000

5	-3.873	-1.084	288.000
	-4.370	-0.260	288.000
	-4.838	0,37847	288.000
10	-5.280	1.331	288.000
	-5.695	2.096	288.000
15	-6.084	2.840	288.000
	-6.448	3.560	288.000
20	-6.788	4.257	288.000
	-7.103	4.929	288.000
25	-7.396	5.576	288.000
	-7.654	6.168	288.000
30	-7.877	6.705	288.000
	-8.068	7.184	288.000
35	-8.240	7.637	288.000
	-8.380	8.033	288.000
40	-8.480	8.340	288.000
	-8.547	8.589	288.000
	-8.585	8.779	288.000
45	-8.597	8.924	288.000
	-8.591	9.004	288.000
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-8.575	9.053	288.000
-8.559	9.074	288.000
-8.549	9.082	288.000

**[0039]** At the same time, each blade 10 therefore has an aerodynamic profile which allows a high conversion efficiency and a high useful life to be maintained.

**[0040]** Furthermore, the aerodynamic profile of the blade 10 according to an embodiment of the invention is obtained with the values of Table I by piling up the series of closed curves and grouping them so as to obtain a continuous aerodynamic profile.

**[0041]** In order to take into account the dimensional variability of each blade 10, the profile of each blade 10 can have a tolerance of +/- 2 mm in a normal direction with respect to the profile of the blade 10 itself.

**[0042]** The profile of each blade 10 can also comprise a coating, applied subsequently and which varies the profile itself.



[0043] Said antiwear coating preferably has a thickness defined in a normal direction at each surface of the blade 10 and ranging from 0 to 0.5 mm.

[0044] It is evident, moreover, that the values of the coordinates of Table I can be multiplied or divided by a corrective constant to obtain a profile in a greater or smaller scale, maintaining the same form.

[0045] According to another aspect of the present invention, a rotor of a ninth phase of a compressor is provided, which comprises a series of blades 10 of the type described above, each of which having a shaped aerodynamic profile, which are fixed to an outer surface of said rotor so as to be uniformly distanced thereon, and also oriented so as to confer a high efficiency to the compressor in which said rotor is preferably inserted.

[0046] According to another aspect of the present invention, a compressor is provided, comprising a rotor of the type described above.

[0047] It can thus be seen that a blade of a rotor of a ninth phase of a compressor according to various embodiments of the present invention achieves the objectives specified above.

[0048] The rotor blade of a ninth phase of a compressor of various embodiments of the present invention thus conceived, can undergo numerous modifications and variants, all included in the same inventive concept.

[0049] Furthermore, in practice, the materials used, as also the dimensions and components, can vary according to technical requirements.

## Claims

1. A blade (10) of a rotor of a ninth phase of a compressor, which can be defined by coordinates of a discreet combination of points, in a Cartesian reference system (X, Y, Z), wherein the axis (Z) is a radial axis intersecting the central axis of the compressor, said blade (10) having a profile which can be identified by means of a series of closed intersection curves between the profile itself and planes (X, Y) lying at distances (Z) from the central axis, said blade (10) being **characterized in that** it comprises a thickening (30), substantially parallel to a base portion (12) of the blade (10) itself, fixable to said rotor, said thickening (30) being substantially situated halfway up the blade (10) and being suitable for shifting the natural resonance frequencies of the blade (10) itself outside a functioning frequency range of said rotor.
2. The blade (10) according to claim 1, **characterized in that** it comprises a further thickening, substantially parallel to said base portion (12) and situated in particular close to a free end (14).
3. The blade (10) according to claim 1 or 2, **characterized in that** it comprises a profile which is identified by a first substantially concave surface (3), which is pressurized, and a second substantially convex surface (5) which is in depression and which is opposite to the first, said two surfaces (3, 5) being continuous and joined to each other to form the profile of said blade 10.
4. The blade (10) according to claim 3, **characterized in that** each closed curve has a maximum thickness determined by the maximum distance between said first surface (3) and said second surface (5), said maximum thickness of each closed curve, along the height of the blade 10 in the direction of a free end (14) of the blade (10), first having a decreasing and then an increasing trend, followed again by a decreasing and finally increasing trend, with a discontinuity point of the slope.
5. The blade (10) according to claim 4, **characterized in that** along the height of the blade (10) in the direction of its free end (14), said maximum thickness has a trend according to the following equations, wherein h represents the height of the blade (10), expressed as a percentage of the total height of the blade (10), and wherein Tmax is the maximum adimensionalized thickness relating to the closed curve corresponding to the height:

$$T_{\max} = -34.522 \cdot h^4 + 36.4 \cdot h^3 - 8.4113 \cdot h^2 - 0.7259 \cdot h + 0.9961$$

for height values ranging from 0 to 45%;

$$T_{\max} = -1.3509 \cdot h + 1.4459$$

for a height ranging from 45% to 58%;

$$T_{\max} = 0.2074 \cdot h + 0.5443$$

for a height ranging from 58% to 86%;

$$T_{\max} = 0.9058 \cdot h - 0.0518$$

for a height ranging from 86% to 100%.

6. The blade (10) according to any of the previous claims, **characterized in that** said closed curves are defined according to Table I, whose values, expressed in millimeters, refer to a profile at room temperature.
7. The blade (10) according to any of the previous claims, **characterized in that** the profile of each blade (10) has a tolerance of +/- 2 mm in a normal direction with respect to the profile of the blade 10 itself.
8. The blade (10) according to any of the previous claims, **characterized in that** the profile of each blade (10) comprises an antiwear coating.
9. The blade (10) according to claim 8, **characterized in that** said coating has a thickness ranging from 0 to 0.5 mm.
10. A rotor of a ninth phase of a compressor, **characterized in that** it comprises a series of blades (10) according to any of the claims 1-9.
11. The rotor according to claim 10, **characterized in that** said series of blades (10) is constrained to an outer surface of said rotor and said series of blades (10) is also uniformly distributed thereon in order to maximize the efficiency of the rotor itself.
12. A compressor **characterized in that** it comprises a rotor according to claim 10 or 11.

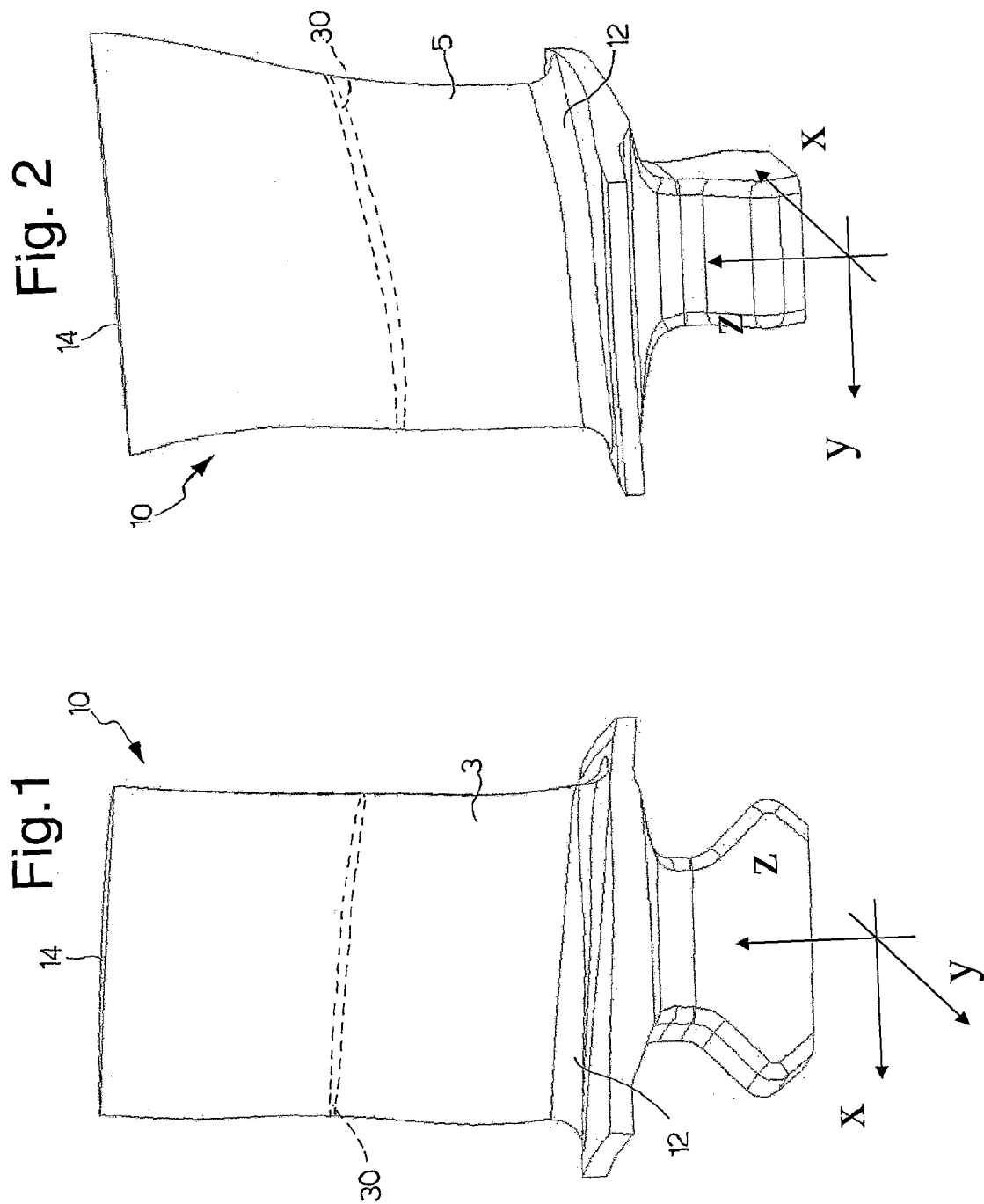


Fig. 3

