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(54) **Spoked spacer for a gas turbine engine**

(57) A spacer (62C) for a gas turbine engine includes a rotor ring (84) defined along an axis of rotation and a plurality of core gas path seals (82) which extend from

the rotor ring (84). Each of the plurality of core gas path seals (82) extends from the rotor ring (84) at an interface (86I), the interface (86I) defined along a spoke (86).

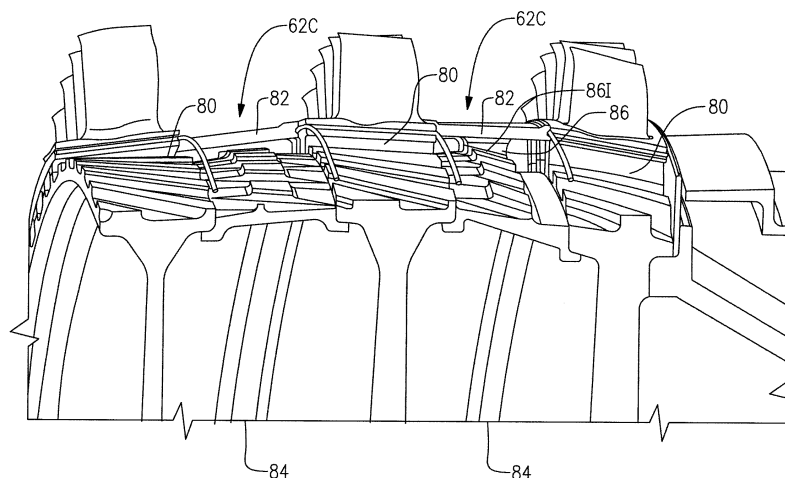


FIG.6

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EUROPEAN SEARCH REPORT

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The present search report has been drawn up for all claims			
Place of search Munich		Date of completion of the search 10 April 2017	Examiner Rapenne, Lionel
CATEGORY OF CITED DOCUMENTS X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document			

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