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(54) **A spacer, a rotor, a spool and a method of orienting a rotor stack load path**

(57) A spacer (62C) for a gas turbine engine includes a rotor ring (84) defined along an axis of rotation. The rotor ring (84) defines a forward circumferential flange (92) which defines a first thickness and an aft circumferential flange (94) which defines a second thickness, the first thickness different than the second thickness. A plurality of core gas path seals (82) which extend from the rotor ring (84) opposite the flanges (92, 94).

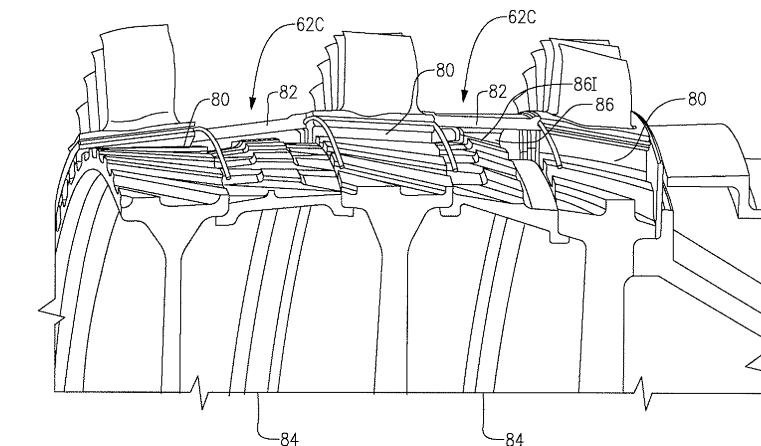


FIG.6

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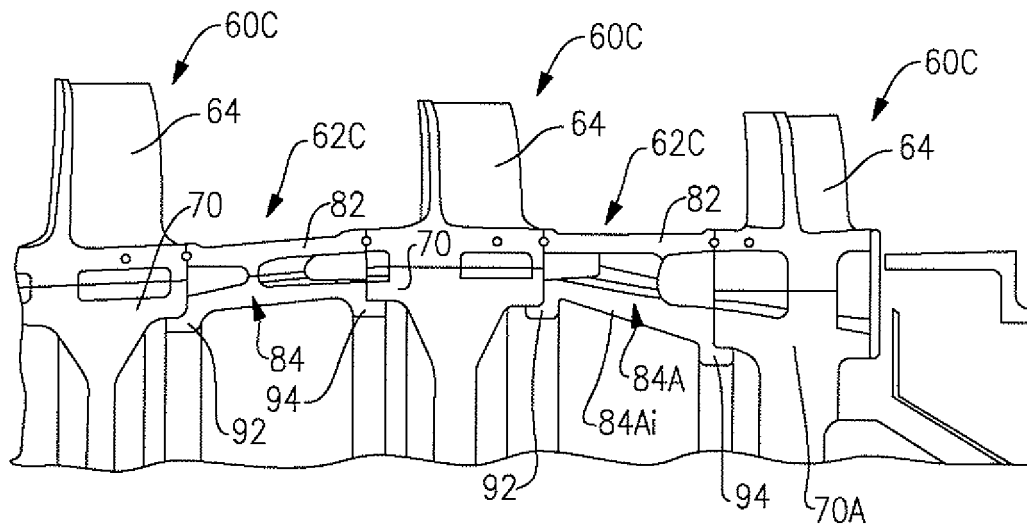


FIG.9



EUROPEAN SEARCH REPORT

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The present search report has been drawn up for all claims			
Place of search Munich		Date of completion of the search 12 April 2017	Examiner Rapenne, Lionel
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**ANNEX TO THE EUROPEAN SEARCH REPORT
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