

EP 2 604 795 A3 (11)

(12)

EUROPEAN PATENT APPLICATION

(88) Date of publication A3: 10.05.2017 Bulletin 2017/19 (51) Int Cl.: F01D 5/18 (2006.01)

F01D 25/12 (2006.01)

(43) Date of publication A2: 19.06.2013 Bulletin 2013/25

(21) Application number: 12194748.5

(22) Date of filing: 29.11.2012

(84) Designated Contracting States:

AL AT BE BG CH CY CZ DE DK EE ES FI FR GB GR HR HU IE IS IT LI LT LU LV MC MK MT NL NO PL PT RO RS SE SI SK SM TR **Designated Extension States: BA ME**

(30) Priority: 15.12.2011 GB 201121531

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(54)Aerofoil blade or vane

(57)An aerofoil blade or vane for the turbine of a gas turbine engine is provided. The blade or vane includes an aerofoil portion which, in use, extends radially across a working gas annulus of the engine. A coolant inlet is formed at an end of the aerofoil portion for entry of a flow of cooling air into the aerofoil portion. A corresponding coolant exhaust is formed at the trailing edge of the aerofoil portion for the flow of spent cooling air from the aerofoil portion. A passage within the aerofoil portion connects the inlet to the exhaust. The blade or vane further includes a fence within the aerofoil portion. The fence extends radially and forwardly from a start position at said end of the aerofoil portion adjacent the trailing edge to an end position. The passage forms a loop which extends along one side of the fence, wraps around the end position, and extends along the other side of the fence to connect the inlet to the exhaust.

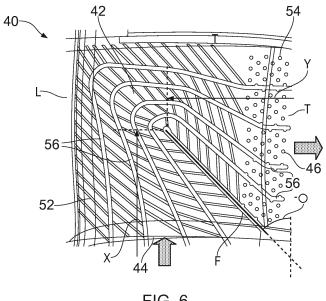


FIG. 6



EUROPEAN SEARCH REPORT

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