

(11) **EP 2 618 059 A3**

(12)

EUROPEAN PATENT APPLICATION

(88) Date of publication A3: 31.12.2014 Bulletin 2015/01

(51) Int Cl.: F23R 3/28^(2006.01) F23D 14/62^(2006.01)

F23R 3/14 (2006.01)

(43) Date of publication A2: **24.07.2013 Bulletin 2013/30**

(21) Application number: 13151601.5

(22) Date of filing: 17.01.2013

(84) Designated Contracting States:

AL AT BE BG CH CY CZ DE DK EE ES FI FR GB GR HR HU IE IS IT LI LT LU LV MC MK MT NL NO PL PT RO RS SE SI SK SM TR Designated Extension States:

BA ME

(30) Priority: 18.01.2012 US 201213352814

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(54) Combustor nozzle/premixer with curved sections

(57) An apparatus for premixing fuel and air prior to combustion in a gas turbine engine includes an annular fuel passage (62) receiving fuel from a fuel source. The annular fuel passage (62) includes a fuel plenum (68) and has a wave shape with a peak section surrounded by trough sections. An annular air passage (70) sur-

rounds the annular fuel passage (62). The annular air passage (70) receives air to be mixed with the fuel in the fuel passage (62) for downstream combustion. A plurality of swirler vanes (72) are disposed in the annular air passage (70) adjacent the fuel plenum (68).

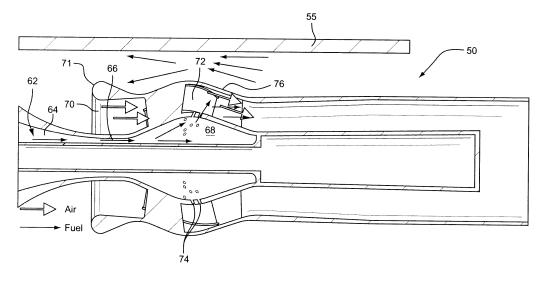


Figure 2

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EUROPEAN SEARCH REPORT

Application Number EP 13 15 1601

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					TECHNICAL FIELDS	
					F23R	
					F23D	
	The present search report has	oeen drawn up for all claims				
	Place of search	Date of completion of	the search		Examiner	
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CATEGORY OF CITED DOCUMENTS X: particularly relevant if taken alone Y: particularly relevant if combined with anoth document of the same category A: technological background		E : ear afte ner D : doo L : doo	vention hed on, or			
O:non	-written disclosure rmediate document	& : member of the same patent family, corresponding document				

ANNEX TO THE EUROPEAN SEARCH REPORT ON EUROPEAN PATENT APPLICATION NO.

EP 13 15 1601

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