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(54) Compressor of a gas turbine system

(57) A compressor 12 of a gas turbine system 10 includes at least one inlet row having a plurality of inlet guide vanes. Also included is at least one stator row 14 having a plurality of stator vanes. Further included is a first actuation mechanism 32 operably connected to the at least one stator row 14, wherein the first actuation

mechanism is configured to positionally manipulate the at least one stator row 14. Yet further included is a second actuation mechanism 34 operably connected to the at least one stator row, wherein the second actuation mechanism is configured to positionally manipulate the at least one stator row.

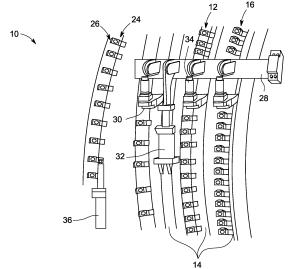


FIG. 1

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